TECHNICAL MANUAL

PHASED MAINTENANCE INSPECTION CHECKLIST FOR

ARMY AH-64D HELICOPTER

Note: This Technical Manual supports AH-64D aircraft serial numbers 96-5001 through 01-5284

HEADQUARTERS, DEPARTMENT OF THE ARMY 24 MAY 2002

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Dates of issue for original and changed pages are:

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TOTAL NUMBER OF PAGES IN THIS PUBLICATION IS 153, CONSISTING OF THE FOLLOWING:

Page	*Change	Page	*Change
No.	No.	No.	No.
A/B blank 1-1 through 1-	2-230 2-370 2-370 2-410 0	2-52 blank 2-53 through 2-65 2-66 blank 2-67 through 2-71 2-72 blank 2-73 through 2-105 2-106 blank 2-107 through 2-115 2-116 blank 2-117 through 2-123 2-124 blank	0 0 0 0 0 50 50 30

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C., 24 MAY 2002

AH-64D HELICOPTER

PHASED MAINTENANCE CHECKLIST

WARNING

CERTAIN INSPECTIONS ARE MANDATORY SAFETY-OF-FLIGHT REQUIREMENTS, AND THE INSPECTION INTERVALS CANNOT BE EXCEEDED. IN THE EVENT THESE INSPECTIONS CANNOT BE ACCOMPLISHED AT THE SPECIFIED INTERVAL, THE AIRCRAFT CONDITION STATUS SYMBOL WILL BE IMMEDIATELY CHANGED TO A RED X. MANDATORY SAFETY-OF-FLIGHT INSPECTION ITEMS ARE PRINTED IN BOLD FACE TYPE.

NOTE

INSPECTION ITEMS CONTAINED IN THIS MANUAL ARE CONSIDERED THE MINIMUM REQUIREMENTS FOR PERFORMING PHASED MAINTENANCE AND MUST BE PERFORMED. THE CUMULATIVE EFFECTS OF INSPECTION DEFERRALS ARE UNKNOWN AND COULD RESULT IN CATASTROPHIC FAILURE OR INCREASED MAINTENANCE AT A LATER DATE. THEREFORE, THE USE OF SPECIAL LETTERING TO EMPHASIZE MANDATORY SAFETY-OF-FLIGHT INSPECTION ITEMS IS NOT TO BE CONSTRUED AS AUTHORITY FOR DEFERRAL OF OTHER INSPECTIONS.

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SECTION I. GENERAL INFORMATION

PHASED SCHEDULE. This phased maintenance inspection checklist contains requirements for inspection of the AH-64D helicotoer on a phased schedule having a 1000-hour (flight hours) cycle with 250-hour/one year phases. Each requirement included herein is designated for accomplishment at least once, but not more than four times during the 1000-hour cycle.

EXCEEDING THE PHASED SCHEDULE. The phased maintenance inspection intervals designated are the maximum and shall not be exceeded except in actual operational emergencies as explained herein. It is the Commander's responsibility to determine (on an individual aircraft basis) when inspection intervals may be exceeded. For this purpose, operational emergencies are conditions of combat, or conditions of disaster which necessitate flight to evacuate aircraft or personnel. Those inspections annotated by a "**C**" in the Inspect Phase Nos. column, along with the DA Form 2408-18 (Equipment Inspection List), are considered the MINIMUM mandatory combat maintenance inspection requirements for helicopters scheduled for imminent deployment to or stationed in a combat environment. Under no circumstances will two combat maintenance inspections be performed sequentially. When aircraft are operated beyond the normal inspection due time because of such emergency situations, a circled red X status symbol and an appropriate statement (to include authority) must be entered in block 16 and 17 of DA Form 2408-13 (Aircraft Inspection and Maintenance Record) until such time as the inspection is complete. When inspections are delayed to meet emergency requirements, Commanders will assure that the aircraft status symbol reverts to a red X and that delayed inspections are accomplished immediately upon termination of the emergency. When unusual local conditions (utilization, type of mission, personnel, periods of inactivity, environmental conditions, etc.) dictate, it is the prerogative and responsibility of the Maintenance Officer to increase the scope and/or frequency of maintenance or inspection as necessary to ensure safe operation (TM 1-1500-328-23).

MAINTENANCE ACTIVITIES. The inspections prescribed by this checklist will be accomplished at specified phases by Aviation Unit Maintenance (AVUM) activities with assistance of Aviation Intermediate Maintenance (AVIM) and Depot Maintenance activities when required. The inspection of the part/component is visual unless stated otherwise.

LIMITATIONS. The checklist does not contain instructions for repair, adjustment, or other means of rectifying conditions. Neither does it contain special tolerances, limits, or instructions for special troubleshooting to find causes for malfunctions. Such data will be obtained from the latest issue of the aircraft TM 1-1520-Longbow/ Apache IETM.

CHANGEOVER TO THE PHASED MAINTENANCE SYSTEM. Changeover shall be accomplished in accordance with instructions provided in TB 55-1500-337-24 entitled, "Phased Maintenance System for Army Aircraft." The requirements of this TB must be accomplished prior to implementation of Phase 1 inspection requirements specified in this checklist.

PRE-INSPECTION MAINTENANCE TEST FLIGHT (MTF). A pre-inspection MTF to duplicate non-hazardous equipment problems, determine unsatisfactory conditions, determine equipment operation problems, etc., is recommended prior to start of aircraft disassembly for phased maintenance inspection. The decision to perform the pre-inspection MTF, however, shall be the responsibility of the unit Maintenance Officer.

SPECIAL INSPECTIONS, CALENDAR INSPECTIONS, AND LUBRICATION REQUIREMENTS. Special inspections, calendar inspections, and lubrication requirements contained in TM 1-1520-Longbow/Apache and those listed on the aircraft's DA Form 2408-18 shall be reviewed and accomplished in accordance with the "inspection due" requirements specified in those documents.

TIME BETWEEN OVERHAUL (TBO) AND RETIREMENT LIFE ITEMS CHECK. Prior to start of the applicable phased maintenance inspections and lubrication, a check will be made of components and their remaining operating hours prior to removal. The latest issue of the aircraft's TM 1-1520-Longbow/Apache and DA Form 2408-16 shall be referred to for a complete listing of components and their TBO and retirement life.

USING THE PHASED INSPECTION CHECKLIST.

a. A new checklist shall be used each time phased maintenance is due on the aircraft. This checklist is arranged such that it can be separated by area and distributed to the maintenance crew.

- (1) Space is provided on each checklist form for entering the following data:
 - (a) The number of the maintenance inspection being performed.
 - (b) Aircraft serial number.
 - (c) Date of inspection.
 - (d) Total hours. (Block provided for local use.)
- (2) For each inspection item a column is provided for entering the following data:
 - (a) Status of the aircraft as the result of the inspection requirement.
 - (b) Aircraft fault and/or remarks indicated by the inspection requirement.
 - (c) Action taken to correct the fault.
 - (d) Initials of person performing the corrective action.

b. This checklist is formatted to eliminate the requirements to use DA Form 2404 as temporary records during phased inspections. Figures 1 thru 3 show examples of methods used to make entries on the phased maintenance checklist forms and the use of supplemental sheets for continuation purposes. This checklist pertains to all AH-64A helicopters and may, therefore, contain inspection requirements applicable to specific equipment not installed on individual aircraft. When this situation is encountered, those requirements that are not applicable need not be performed.

c. A supplemental Checklist Sheet form (DA Form 4676-R) (figure 3, Sheet 2) provided at the end of Section I of this checklist is to be used for local reproduction. Copies of this form will be used to write up faults, remarks, and corrective actions when additional space is required. These supplemental sheets will be used instead of DA Form 2404 in the accomplishment of the phased maintenance inspections.

d. Faults and remarks on the DA Form 2408-13 and DA Form 2408-14 may be transcribed to this checklist at the discretion of the unit Maintenance Officer.

PHASE NUMBERS. In the column headed "Inspect Phase Nos." and adjacent to the sequence number of each inspection requirement, there will appear the word "ALL" or a series of numbers. The word "ALL" indicates that inspection requirement shall be accomplished at each phase (or at every 250-hour/one year interval) of the 1000-hour cycle. The numbers represent the phase number at which that inspection requirement is to be accomplished. For example, if the numbers 2 and 4 are shown, that inspection requirement is to be accomplished at phases 2 and 4 only (or at 500-hour interval). If only one number is indicated, then that inspection requirement is accomplished at that phase (or at every 1000-hour interval). At the completion of phase 4, the cycle starts over again with Phase 1.

STATUS SYMBOLS. All faults and deficiencies discovered during the inspection will be recorded on DA Form 2408-13-1/2408-13-1-E. The status symbols used are the same as those defined in DA PAM 738-751. The status symbol shall be entered by the person(s) performing the inspection and is determined by the type of fault that is found. Do not enter a horizontal dash (-) on the checksheet merely to show a particular inspection requirement is due. If an inspection reveals no fault, a status symbol will not be entered. The person clearing the fault shall place his last name initial over the status symbol. A red X or a circled red X symbol will not be initialed over until after the corrective action has been approved and signed off by a Technical Inspector or designated supervisor.

FAULTS AND/OR REMARKS. Fault entries in the Faults and/or Remarks column shall be brief remarks which describe the conditions resulting from the inspection and which require corrective action. The initials of the person making the entry will be entered immediately after the entry. If no fault is found, this column will be left blank.

ACTION TAKEN.

a. Entries in the Action Taken column shall be brief remarks which describe the action taken to correct the fault described in the adjacent Faults and/or Remarks column. When faults are assigned a red X status, the corrective action shall be inspected and signed off by the Technical Inspector or designated supervisor.

b. If no fault was found, an appropriate remark shall be entered in this column to indicate that the inspection was accomplished, e. g., "Inspected and found OK." If an inspection item is not applicable to the particular inspection phase number in work or to specific equipment installed on an individual aircraft, a "N/A" entry is required. The initials of the person making the entry shall be entered in the Initial column.

INITIAL. The person correcting the indicated fault shall enter his initials in the initial column opposite the first line of the Action Taken entry.

FINAL RECORDS CHECK. After all corrective actions have been completed and following completion of the phased inspection, the Technical Inspector or designated supervisor shall verify that all applicable forms and records have been properly updated. All uncorrected faults shall be entered on DA Form 2408-13, prepared for that date or to the DA Form 2408-14. A Final Records Checklist (table I) is provided to ensure forms and records have been inspected for completeness and accuracy prior to release of the aircraft from the phased maintenance inspection. The inspector verifying the final records check shall enter his initials adjacent to the indicated form on the Final Records Checklist. The initials entered shall be registered on the Signature Sheet (table II) adjacent to that person's signature.

SIGNATURE SHEET. All personnel performing inspection and/or maintenance tasks shall place their signatures and initials on the signature sheet (table II). The purpose of the signature sheet is to provide a correlation between initials entered on the individual checklist sheets and the actual names of the personnel accomplishing these tasks.

MAINTENANCE OPERATIONAL CHECKS. After the completion of any required corrective actions to any of the components of a functional system of the aircraft, maintenance operational checks (MOC) shall be performed on that system to determine the effectiveness of the maintenance actions performed and to verify the proper operation of that system. These MOC shall be performed in accordance with TM 1-1500-328-23. Copies of supplemental sheets (DA Form 4676-R) may be used to record and sign off the Maintenance Operational Checks performed.

MAINTENANCE TEST FLIGHT. When all required inspections in Section II have been accomplished and initialed in accordance with the above procedure, a daily inspection in accordance with the TM specified in Section II will be performed on the aircraft to permit performance of a maintenance test flight (MTF). The MTF shall be performed in accordance with the requirements of TM 1-1520-Longbow/Apache and TM 1-1500-328-23 using the MTF form in the MTF technical manual. A suggested maintenance test flight checksheet (figure 4) and a rotor smoothing record (figure 5) are provided at the end of Section I.

CHECKLIST DISPOSITION. The completion of each phased maintenance inspection shall be recorded on DA Form 2408-13 and 2408-15 as prescribed by DA PAM 738-751. The signed checklist, together with all continuation sheets, shall be attached to DA Form 2408-13 and filed for the six months period as required by DA PAM 738-751. At the end of the six months period, records will be destroyed per disposition instruction for DA Form 2408-13-1/2408-13-1-E in paragraph 2-9.D.(2) of DA Pamphlet 738-751.

INSPECTION AREAS. Figure 6 reflects the inspection areas of the AH-64D helicopter. Those areas are titled as shown. Figure 7 shows the location of access doors and panels which require removal at various phased maintenance inspections.

REPORTING ERRORS AND RECOMMENDING IMPROVEMENTS. You can help improve this manual. If you find any mistakes or if you know of a way to improve the procedures, please let us know. Mail your letter, DA Form 2028 (Recommended Changes to Publications and Blank Forms), or DA Form 2028-2 located in the back of this manual direct to: Commander, U.S. Army Aviation and Missile Command, ATTN: AMSAM-MMC-MA-NP, Redstone Arsenal, AL, 35898-5230. A reply will be furnished to you.

РНА	SE	NO1 P	HASE	D MAINTENANCE CHECKLIST			
L	_EFT	Area Name and No. FORWARD AVIONICS BAY AND MLG		Aircraft Serial No. 77-23259	APR 81	Total Hrs. This Area 510	l
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL C 👞	1.	MLG wheel for cracks, distortion, or corrosion. Hub for grease leakage. Tire for INDICATES COMBAT SITUATION INSPECT	ON			Insp – OK	WCB
ALL C	2.	MLG wheel brake for fluid leakage, cracked housing or corrosion. Check brake pucks for wear.	ß	corrosion on bottom of brake housing WCB Leakage from housing inlet connection WCB (continued on Supplemen	Inlet fit	ting tightened	WCB WCB
2,4	3.	Search light for corrosion, loose or missing fasteners, and security. Lens for cracks or evidence of overheating. Wiring for loose connections. Chafing, deterioration, and security.			HEAVY SEPAR A BLO	fictitious and are intended	

Figure 1. Example of Phased Maintenance Checklist Title Sheet

РНА	SE NO		a Name a RANSM	and No. IISSION - 9		Aircraft Serial No. 77–23259	APR 81
Inspect Phase Nos.	Inspection Rec	quirements	Status	Faults and/or Rema	arks	Action Taken	Initial
2,4	5. Breathers cleaned.				_ EXAM	PLE Insp-OK	WCB
	Access L200, R200				-EXA-		
ALL	6. Lube oil and filters ch	anged.		T	WO PEOP	Insp – OK LE PERFORMED THIS	G WCB
	Access L200, R200				NSPECTIO NITIALED.	N. BOTH HAVE	
ALL	 Lube oil level sight ga leakage, and security 	•	B		HAT THE S	<i>Tightened</i> AME PERSON (WCB) CORRECTED THE	WCB
	Access L200, R200			FAULT.			
2,4	8. Input shafts and coup distortion, and corros	blings for cracks, dents, ion.	<u> </u>	No.2 shaft inp Coupling diaphr cracked W(a M	Coupling replaced QA INITIALS ON LAS	
	Access L200, R200, I	LN6, RN6		QA SIGN		Insp-OK Harold S.	Smith HSS
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	For use o	PHASED MAINTENANCE CH f this form, see TM 55-1510 series and TM 55-1520 series, the p			Readiness Command.	DATE 4 APR 8	\$1
PHASE	no. 1	AREA NAME AND NUMBER	.G - 3	AIRCRAFT SERIAL NO. 77-23259	TOTAL HOURS THIS AR		•
INSPECT AREA NO.	INSPECT ITEM NO	INSPECTION REQUIREMENTS	STATUS	FAULTS AND/OR REMARKS	ACTION TAKEN		INITIAL
3	9	(continued)		Brake puck worn, l unserviceable wcb	Replaced		WCB
		A SUPPLEMENTAL SHEET IS USED WHEN SPACE IS NOT AVAILABLE ON CHECKLIST PAGE FOR ALL FAULTS OR CORRECTIVE ACTION.					
		Check work area for tool		DD REMINDER " s after completion of maintenance and Inspection.			

DA Form 4676-R 1 Dec 77

Figure 3. Example of Checklist Supplemental Sheet (Sheet 1 of 2)

	For use of	PHASED MAINTENANCE this form, see TM 55-1510 series and TM 55-1520 series,	the proponent a	T (SUPPLEMENTAL SHEET) agency is the US Army Materiel Development an	d Readiness Command.	DATE
PHASE		AREA NAME AND NUMBER		AIRCRAFT SERIAL NO.	TOTAL HOURS THIS ARE	A
INSPECT AREA NO.	INSPECT ITEM NO	INSPECTION REQUIREMENTS	STATUS	FAULTS AND/OR REMARKS	ACTION TAKEN	INITIAL
		Check work area fo		D REMINDER" after completion of maintenance and Inspection	۱.	

DA Form 4676-R 1 Dec 77

Figure 3. Example of Checklist Supplemental Sheet (Sheet 2 of 2)

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					LEVERS	POWER LEVERS	.≓
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						PNVS	.00
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SEC		c. TIME TO IDLE				a. GEN 1	
% N _G	JUT	b. STARTER DROPOUT			GENERATOR SYSTEM	GENERAI	.
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		ENG 2 START	сі		RT	APU START	.
ိင		f. TGT			STARTING APU - PILOT/CPG	STARTING	
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Figure 4. Example of Maintenance Test Flight Checksheet (Sheet 1 of 3)

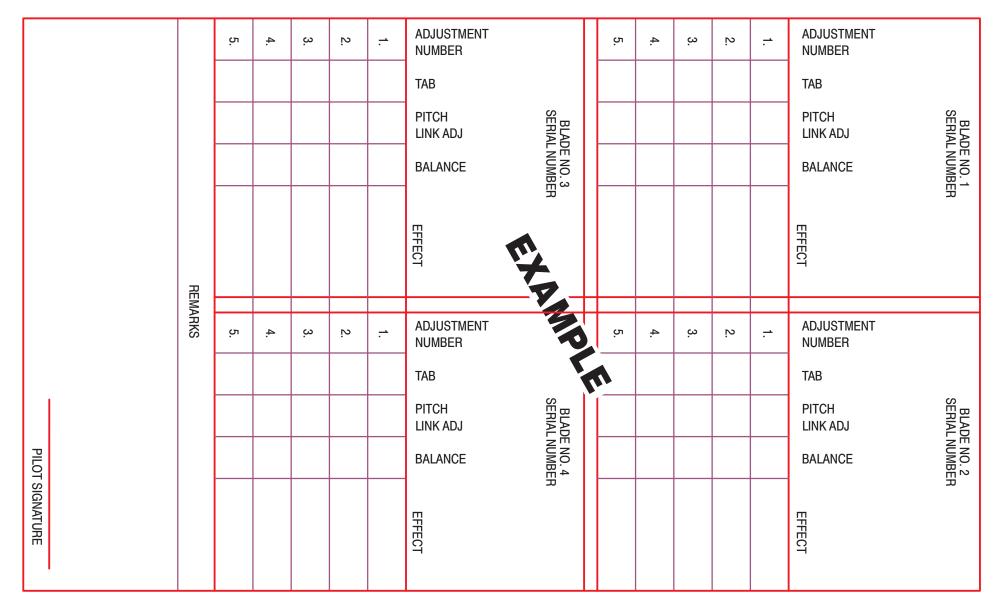
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ENGINE RUNUPFLIGHT CHECK1"G" SPRING121. TAKE OFFa. TORQUE%%2. CRUISEENGINE CHOP CIRCUIT3. AUTOROTATIONa. ENG1a. PRESSALTa. ENG1b. FATb. ENG2.C. NrECU LOCK OUT4.ATTITUDE HOLDa. ENG5.MANEUVERING FLIGHTb. ENG4.ATTITUDE HOLDa. ENG5.MANEUVERING FLIGHTb. ENG6.POS/NEG "G" CHECKFUEL SYSTEM16.DEK -FD/LS5.A. TORQUEa. XMN1PSI<°C	%	%	d. TORQUE		%	8			
ENGINE RUNUPFLIGHT CHECK1'G'' SPRING12I.a. TORQUE%%2CRUISEENGINE CHOP CIRCUITIa. AUTOROTATIONENGINE CHOP CIRCUTIa. PRESS ALTENGINE OVERSPEED TESTIb. FATa. ENG 1b. ENG 2.Ia. ENG 1Ib. FATb. ENG 2I.b. ENG 2IS. MANEUVERING FLIGHTb. ENG 2IS. MANEUVERING FLIGHTb. ENG 2S''V. H FLIGHTb. ENG 2PSI<'C	°C	റ	c. TGT		PSI	IS			
ENGINE RINUPFLIGHT CHECK1121.a. TORQUE%2.CRUISEENGINE CHOP CIRCUIT3.AUTOROTATIONENGINE CHOP CIRCUIT3.AUTOROTATIONENGINE OVERSPEED TEST4A. DRESS ALTa. ENG 1b. FATb. FATb. ENG 2c. NrECU LOCK OUT5.A. ENG 16.b. ENG 27.VH FLIGHT5.b. ENG 27.VH FLIGHTb. ENG 27.VH FLIGHTb. ENG 28.C. NGB 1PSI°C1.A. NG%NGB1.A. NG%NG1.A. NG%MANA. NG%A. NG%MANA. NG%MANA. NG%MANA. NG%MANA. NG%A. N	2	-	ENG		ငိ	C			
ENGINE RIVUPFLIGHT CHECK1 "G" SPRING121. TAKE OFFa. TORQUE%2. CRUISE2. CRUISEENGINE CHOP CIRCUIT3. AUTOROTATION3. AUTOROTATIONENGINE CHOP CIRCUT3. AUTOROTATIONa. ENG 13. AUTOROTATIONb. ENG 25. FATb. ENG 14. ATTITUDE HOLDa. ENG 15. MANEUVERING FLIGHTb. ENG 24. ATTITUDE HOLDa. ENG 15. MANEUVERING FLIGHTb. ENG 27. VH FLIGHTb. ENG 28. TOROUEDEK - FD/LS6. POSINEG "G" CHECKDEK - FD/LS7. VH FLIGHTa. XMN 1PSI<°C	ငိ		1.		%	8	N _r		
ENGINE RUNUPFLIGHT CHECK1121. TAKE OFFa. TORQUE%2. CRUISEENGINE CHOP CIRCUIT3. AUTOROTATIONENGINE CHOP CIRCUIT3. AUTOROTATIONENGINE CHOP CIRCUIT3. AUTOROTATIONa. ENG 13. AUTOROTATIONb. ENG 24. ATTITUDE HOLDa. ENG 15. MANEUVERING FLIGHTb. ENG 24. ATTITUDE HOLDa. ENG 15. MANEUVERING FLIGHTb. ENG 25. MANEUVERING FLIGHTb. ENG 27. VH FLIGHTDEK - FD/LS6. POS/NEG "G" CHECKb. XMN 1PSI°C6. AISc. NGB 1PSI°C6. ALSc. NGB 1PSI°C6. TASeNG1A. NG%8. ENGINE PERFORMANCE	피		a. PRESS ALT		%	8	Np		
ENGINE RUNUPFLIGHT CHECK1 "G" SPRING12FLIGHT CHECKa. TORQUE%%2.CRUISEENGINE CHOP CIRCUIT3.AUTOROTATIONENGINE CHOP CIRCUIT3.AUTOROTATIONENGINE OVERSPEED TEST4.AITTOROTATIONa. ENG 15.FATb. ENG 26.POS/NEG "G" CHECKc. ULOCK OUT4.ATTTUDE HOLDa. ENG 15.MANEUVERING FLIGHTb. ENG 25.MANEUVERING FLIGHTb. ENG 27.V _H FLIGHTb. ENG 2S.NANEUVERING FLIGHTb. ENG 2PSI°Cb. TGTa. XMN 1PSI°Cc. NGb. XMN 2PSI°Cc. NGb. XMN 2PSI°Cc. TASc. NGB 1PSI°Cc. TASd. NGB 2PSI°Cc. TASeNG12g. FAT		NANCE	ENGINE PERFORM	.œ	%	8	. N _G		
ENGINE RUNUPFLIGHT CHECK1'G'' SPRING121.TAKE OFFa. TORQUE%%2.CRUISEENGINE CHOP CIRCUIT%3.AUTOROTATIONENGINE OVERSPEED TESTA.NATTOROTATIONa. ENG 1b. ENG 2ECU LOCK OUTa. ENG 1b. ENG 2b. ENG 2c. NGB 1<	ိင		g. FAT		2	-		Ś	
ENGINE RUNUPFLIGHT CHECK1"G" SPRING121.TAKE OFFa. TORQUE%%2.CRUISEENGINE CHOP CIRCUIT3.AUTOROTATIONENGINE OVERSPEED TEST3.AUTOROTATIONa. ENG 1b. ENG 2a. ENG 1b. ENG 2a. ENG 1b. ENG 2b. ENG 2c. NG 2DEK - FD/LSb. XMN 2b. TASc. NGB 1c. NGB 1c. NG 2	ㅋ		f. PRESS ALT		ိငိ	ISc			
ENGINE RUNUPFLIGHT CHECK1 "G" SPRING121. TAKE OFFa. TORQUE%%2. CRUISEENGINE CHOP CIRCUIT3. AUTOROTATION8. AUTOROTATIONENGINE OVERSPEED TEST3. AUTOROTATIONa. ENG 1b. ENG 2c. Nrb. ENG 2a. ENG 1b. ENG 2b. ENG 2b. ENG 2b. ENG 2c. Nrb. ENG 2b. ENG 1b. ENG 2c. Nrb. ENG 2b. ENG 2c. Nrb. ENG 2c. Nn EUVERING FLIGHTb. ENG 2b. ENG 2c. Nn EUVERING FLIGHTb. ENG 2b. ENG 2c. Ng HLIGHTb. EFORE TAXI CHECKDEK - FD/LSb. XMN 1PSIc. Ngb. XMN 2PSIc. Ngd. IAS	KNOTS		e. TAS		ငိ	ISc			
ENGINE RUNUPFLIGHT CHECK1 $^{r}G^{r}$ SPRING121.a. TORQUE $^{\circ}G^{r}$ $^{\circ}G^{r}$ 2.CRUISEENGINE CHOP CIRCUIT13.AUTOROTATIONENGINE OVERSPEED TEST13.AUTOROTATIONa. ENG 1b. ENG 2ECU LOCK OUTa. ENG 1b. ENG 2b. ENG 2puel systemPUEL SYSTEMDEK - FD/LSa. XMN 1PSI°CA. XMN 1	KNOTS		d. IAS		°C	ISc	XMN 2		
ENGINE RUNUPFLIGHT CHECK1 $"G"$ SPRING121. TAKE OFFa. TORQUE $"S"$ $"S"$ 2. CRUISEENGINE CHOP CIRCUIT $"S"$ $"S"$ $"S"$ ENGINE OVERSPEED TEST $"S"$ $"S"$ $"S"$ a. ENG 1 $"S"$ $"S"$ $"S"$ b. ENG 2 $"S"$ $"S"$ $"S"$ a. ENG 1 $"S"$ $"S"$ $"S"$ a. ENG 1 $"S"$ $"S"$ $"S"$ b. ENG 2 $"S"$ $"S"$ $"SNEEN FULGHTb. ENG 2"S""SNEEN FULGHT"S"b. ENG 2"S""SNEEG "G" CHECKFUEL SYSTEM"S""S""S"BEFORE TAXI CHECK"S""S""S"DEK - FD/LSS"S""S"$	%				ငိ	<u>IS</u>	XMN 1		
ENGINE RUNUPFLIGHT CHECK1'G" SPRING12I. TAKE OFFa. TORQUE'S'SCRUISEENGINE CHOP CIRCUIT'SAUTOROTATIONENGINE OVERSPEED TESTSAUTOROTATIONa. ENG 1'S'Sb. ENG 2'S'SECU LOCK OUTS'Sa. ENG 1'S'SA. ENG 1'S'SMANEUVERING FLIGHTSMANEUVERING FLIGHTb. ENG 2'S'SPUEL SYSTEMS'SBEFORE TAXI CHECKS'AA. TORQUES'A	ိင						DEK – FD/LS	. ``	
FLIGHT CHECK1 "G" SPRING12I. TAKE OFFa. TORQUE%%2. CRUISEENGINE CHOP CIRCUIT3. AUTOROTATIONENGINE OVERSPEED TEST3. AUTOROTATIONa. ENG 1b. FATb. ENG 2c. NrECU LOCK OUT4. ATTITUDE HOLDa. ENG 15. MANEUVERING FLIGHTb. ENG 25. MANEUVERING FLIGHTb. ENG 27. VH FLIGHT	%		a. TORQUE				BEFORE TAXI CHECK		
FLIGHT CHECK1 "G" SPRING12I. TAKE OFFa. TORQUE%%2. CRUISEENGINE CHOP CIRCUIT%3. AUTOROTATIONENGINE OVERSPEED TEST%3. AUTOROTATIONa. ENG 1b. FATb. FATb. ENG 2c. Nra. ENG 1c. Nra. ENG 14. ATTITUDE HOLDa. ENG 15. MANEUVERING FLIGHTb. ENG 25. POS/NEG "G" CHECK			V _H FLIGHT	7.			FUEL SYSTEM	ςī	
ENGINE RUNUPFLIGHT CHECK1 "G" SPRING121. TAKE OFFa. TORQUE%%2. CRUISEENGINE CHOP CIRCUIT3. AUTOROTATION3. AUTOROTATIONENGINE OVERSPEED TEST4. SATRESS ALTa. ENG 1b. FATb. ENG 2c. NrECU LOCK OUT4. ATTITUDE HOLDa. ENG 15. MANEUVERING FLIGHT		CK	POS/NEG "G" CHE	6.			b. ENG 2		
ENGINE RUNUPFLIGHT CHECK1 "G" SPRING121. TAKE OFFa. TORQUE%%2. CRUISEENGINE CHOP CIRCUIT3. AUTOROTATION3. AUTOROTATIONENGINE OVERSPEED TEST4. ATTITUDE HOLDb. ENG 25. FATb. ENG 25. ATTITUDE HOLD		JGHT	MANEUVERING FL	'n			a. ENG 1		
ENGINE RUNUPFLIGHT CHECK1 "G" SPRING121. TAKE OFFa. TORQUE%%2. CRUISEENGINE CHOP CIRCUIT3. AUTOROTATIONENGINE OVERSPEED TEST3. APRESS ALTa. ENG 1b. FATb. ENG 2c. N_r			ATTITUDE HOLD	. 4			ECU LOCK OUT	.4	
ENGINE RUNUPFLIGHT CHECK1 "G" SPRING121. TAKE OFFa. TORQUE%%2. CRUISEENGINE CHOP CIRCUIT3. AUTOROTATION3. AUTOROTATIONENGINE OVERSPEED TEST4. PRESS ALTa. ENG 1b. FAT	%		c. N _r				b. ENG 2		
ENGINE RUNUP FLIGHT CHECK 1 "G" SPRING 1 2 1. TAKE OFF a. TORQUE % % 2. CRUISE ENGINE CHOP CIRCUIT 3. AUTOROTATION 3. PRESS ALT	ဂိ						a. ENG 1		
ENGINE RUNUP 1 2 1. TAKE C 1 "G" SPRING 1 2 1. TAKE C a. TORQUE % % 2. CRUISI ENGINE CHOP CIRCUIT 3. AUTOR	ㅋ		a. PRESS ALT			1 T	ENGINE OVERSPEED TES	ω	
ENGINE RUNUP 1 2 1. TAKE C 1 "G" SPRING 1 2 1. TAKE C a. TORQUE % % 2. CRUISI			AUTOROTATION	<u>.</u> ω			ENGINE CHOP CIRCUIT	ы	
ENGINE RUNUP 1 "G" SPRING 1 2 1. TAKE C			CRUISE	ы	%	%			
			TAKE OFF	. ^	N			. ``	
		IECK	FLIGHT CH				ENGINE RUNUP		

Figure 4. Example of Maintenance Test Flight Checksheet (Sheet 2 of 3)



Figure 4. Example of Maintenance Test Flight Checksheet (Sheet 3 of 3)





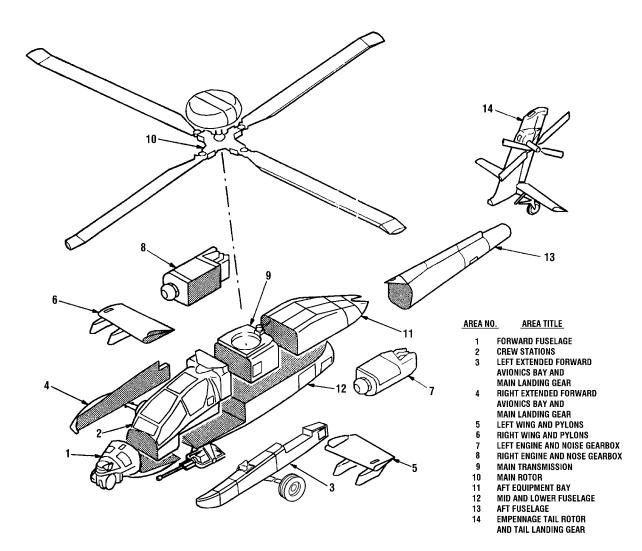
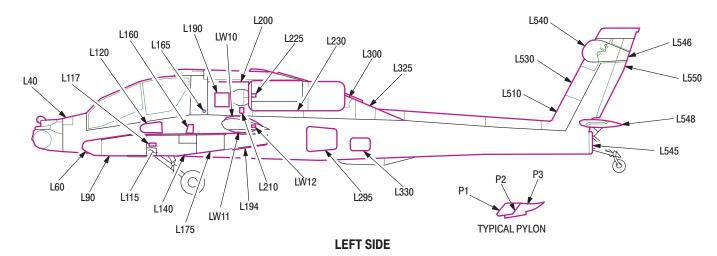
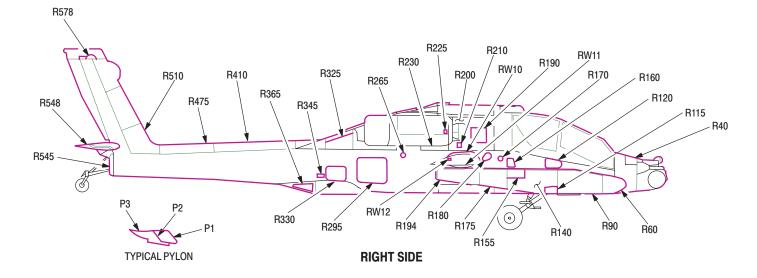


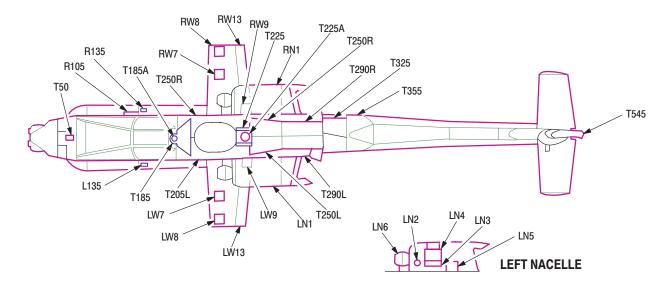
Figure 6. Inspection Area Diagram

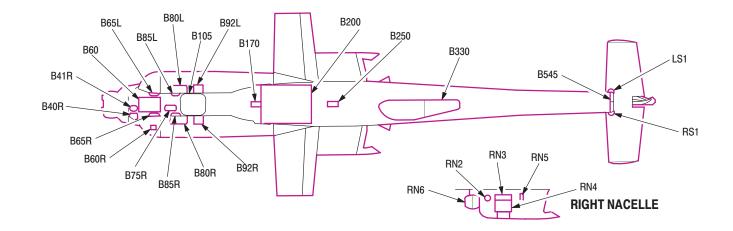




M06-003-1A

Figure 7. Inspection Access Provisions (Sheet 1 of 2)





M06-003-2A

Figure 7. Inspection Access Provisions (Sheet 2 of 2)

Table I. Final Records Checklist

This checklist is provided to ensure that the indicated forms and records have been inspected for presence, completeness, legibility and accuracy prior to releasing the aircraft from a phased inspection. Verification of inspection will be indicated by placing the initials of the inspector in the appropriate initial block.

AIRCRAFT LOG BOOK	INITIAL	HISTORICAL RECORDS	INITIAL
DA FORM 2408		DA FORM 2408-5	
DA FORM 2408-12			
DA FORM 2408-13			
DA FORM 2408-14		DA FORM 2408-9	
DA FORM 2408-18		DA FORM 2408-15	
TM 1-1520-251-PMS		DA FORM 2408-16	
		DA FORM 2408-17	
TM 55-1520-251-MTF		DA FORM 2408-19	
LOCALLY REQUIRED FORMS		LOCALLY REQUIRED FORMS	

PRODUCTION CONTROL RECORDS	INITIAL	QUALITY CONTROL	INITIAL
FLOW CHART		TBO FILE	
STATUS BOARD		QA FILE	
WORK ORDER FILE		SERIAL NUMBER FILE	
MWO FILE		AOAP FILE	
CONFIGURATION CHART		INVENTORY RECORDS	
2405 LOG		WEIGHT AND BALANCE	
1352 REPORTS		MSG FILE	
LOCAL RECORDS		DA FORM 2410 SUBMITTED	
		LOCAL RECORDS	

Table II. Signature Sheet

Signature of Person Accomplishing Necessary Work	-	Initial
Signature of Person Accomplishing Necessary Work	-	Initial
Signature of Person Accomplishing Necessary Work	-	Initial
Signature of Person Accomplishing Necessary Work	-	Initial
Signature of Person Accomplishing Necessary Work	-	Initial
Signature of Person Accomplishing Necessary Work	-	Initial
	-	Initial
	-	Initial
Signature of Maintenance Supervisor	-	Initial
Signature of Technical Inspector	-	Initial
Signature of Maintenance Officer	-	Initial

Table III. Signature Sheet

Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
	Initial
	Initial
Signature of Maintenance Supervisor	 Initial
Signature of Technical Inspector	 Initial
Signature of Maintenance Officer	 Initial

SECTION II. INSPECTION CHECKLIST

WARNING

ACCIDENTAL ACTUATION OF HELICOPTER POWER PLANT, HYDRAULIC SYSTEM, CANOPY JETTISON SYSTEM, OR FIRING OF ARMAMENT OR STORES JETTISON BALLISTICS MAY CAUSE SEVERE INJURY OR DEATH. BEFORE STARTING INSPECTION, HELICOPTER SAFETY CHECK MUST BE PERFORMED (TM 1-1520-LONGBOW/APACHE) AND ALL ARMAMENT MUST BE SAFETIED, DEACTIVATED, AND CLEARED (TM 9-1090-208-23 AND TM 9-1427-475-23).

NOTE

Prior to the start of the phased maintenance inspection, it is recommended that a pre-inspection Maintenance Test Flight (MTF) be conducted. Accomplishment of the MTF shall be determined by the unit maintenance officer. The pre-inspection MTF should be conducted by a maintenance test pilot following a review of the aircraft forms and records and a briefing from the crew of the helicopter. The MTF is recommended to assess the helicopter performance and identify deficiencies that should be corrected while the helicopter is undergoing phased maintenance inspections.

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspection.

РНА	SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
	_	Area Name and No. GENERAL		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL C	1.	Prior to inspection, check forms and records for record deficiencies.					
ALL C	2.	Perform engine run-up and check for proper engine operation per TM-1-1520-251-MTF.					
ALL	3.	Perform APU start and MOC per IETM.					
ALL C	4.	Take oil samples (engine no. 1 and no.2, nose gearboxes, main transmission, and APU) within 30 minutes of engine shutdown.					
ALL C	5.	Fuel tanks will be fully serviiced prior to start of phased inspection. If maintenance is to be accomplished which requires defueling, this item may be deferred until after such maintenance is completed.					
ALL	6.	Depanel aircraft.					

РНА	SE N	0 P	HASEI	D MAINTENANCE CHECKLIST			
		Area Name and No. GENERAL		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	7. I	Remove gun turret.					
ALL	8. I	Remove ammo magazine.					
ALL	9. \	Wash aircraft.					
ALL	10. I	Remove main rotor blades.					
ALL C		Remove retention nut (send retention nut to production control for further disposition).					
ALL	12. I	Remove pilot seat.					
ALL	13. I	Remove CPG seat.					
ALL		Remove pylon ejector cartridges and stow IAW unit SOP.					

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspection.

PHA	SE	NO P	HASED M	AINTENANCE CHECKLIST			
		Area Name and No. FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Are	a
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	A	ction Taken	Initial
ALL	1.	(WARNING Do not attempt to move jettison handle).					
		Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing attaching hardware.					
ALL	2.	Access panels, fairings, and doors for deformations, cracks, corrosion, loose or missing hardware. Door and hinges for damage, binding, and security. Latches for security and proper operation. Check for worn or deteriorated seals and copper coat.					
ALL	3.	Interior components for mounting security and loose or missing hardware.					
ALL	4.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.					

PHA	SE	NO F	PHASED M	AINTENANCE CHECKLIST			
		Area Name and No. FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Are	ea
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	A	ction Taken	Initial
ALL	5.	Avionics Equipment for proper storage, external damage, and loose connectors. Wiring harness for chafing and deterioration.					
ALL	6.	CPG brake master cylinders for leakage, cracks, and loose or missing hardware. None allowed, if found replace with undamaged like item. Hydraulic lines for leakage, chafing and connection security.					
2,4 C	7.	Flight control rods for dents, cracks, corrosion, security and evidence of interference. Rod ends for worn and seized bearings.					
2,4 C	8.	Flight control bellcranks for cracks, corrosion, security and evidence of interference. Check pivot bearings for looseness.					

РНА	SE	NO P	HASED	D MAINTENANCE CHECKLIST			
		Area Name and No. FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	l
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
2,4 C	9.	CPG Decoupler (ARDD) units for cracks, corrosion, security, and evidence of interference. Check for looseness and lost motion.					
2,4 C	10.	LVDT (S) for cracks, corrosion, and mounting security. Wiring harnesses for loose connections, chafing, or deterioration and evidence of interference.					
ALL C	11.	Exterior canopy jettison components for cut or broken transfer tube, bulged, chafed or swollen union, loose or missing hardware and safety wire torn or missing streamer.					
ALL	12.	Canopy emergency release drain hose and outlet for breaks.					

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
	ARM	Area Name and No. MAMENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Aircraft interface assembly and fairings for cracks, distortion, missing fastners and damage.					
ALL	2.	Aircraft Interface Assembly wiring harnesses for chafed or broken wires and loose, bent, burned or broken pins or sockets.					
ALL	3.	Glass gaskets on both dayside and nightside sensor assemblies for damage.					
ALL	4.	AIA bonding strap assembly for loose, damaged and frayed wires. Check bonding strap assembly mounting hardware for security and damage.					
ALL	5.	Area around lower support laser shield for cracked or peeling paint.					
ALL	6.	Boresight assembly optics for contamination.					

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspection.

РНА	SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
	ARM	Area Name and No. MAMENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	7.	Area around boresight assembly for cracked, broken, or peeling paint.					
ALL	8.	ECA for external damage.					
ALL	9.	Remove and replace ECS air filter assembly.					
ALL	10.	ECS molded hose for cracks, cuts, deterioration, and loose fit.					
ALL	11.	Check external surfaces of night sensor shroud for dents, cracks, punctures, and window for cracks or chips.					
ALL	12.	Remove night sensor shroud assembly. Inspect inner surface of window for contamination, pitting, nicks and scratches.					

РНА	ASE NO P	PHASEI	D MAINTENANCE CHECKLIST			
	Area Name and No. ARMAMENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	 Internal painted surfaces for scratched, chipped, or peeled paint. Check for any damage to components. Relate scratches and chips to possible points of contacts. 					
ALL	 NSA harnesses for chafed or broken wires and loose or corroded connections. Anti-ice CCA A1 and connector P1 for loose mounting hardware. Window temperature sensor RT1 and thermostat S2 for bonding separation. 					
ALL	15. NSA optics for contamination, pitting, nicks, scratches and peeling of coating.					
ALL	16. Move TADS turret as required. External surfaces of day shroud for dents, cracks or punctures and window for chips or cracks.					
ALL	 Remove day sensor shroud assembly. Inspect innner surfaces of window for contamination, pitting, nicks, scratches and peeling of coating. 					

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspection.

РНА	SE N	0 P	HASED	MAINTENANCE CHECKLIST			
	ARMA	Area Name and No. MENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL		nternal painted surfaces for scratched, chipped, or peeled paint. Check for any damage to components. Relate scratches and chips to possible points of contact.					
ALL	l (r s	Harnesses for chafed or broken wires and oose or corroded connections. Anti-ice CCA A1 and connector P1 for loose mounting hardware. Window temperature sensor RT1 and thermostat S1 for bonding separation.					
ALL		DSA optics for contamination, nicks, scratches, and peeling of optical coating.					
ALL	f	PNVS shroud assembly external surfaces for dents, cracks, chips, punctures, scratches, peeled paint, and window for chips or cracks.					
ALL	\ \	Remove PNVS shroud. Inner surface of window for contamination, pitting, nicks, scratches, chips, and peeled paint.					

РНА	SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
	ARN	Area Name and No. MAMENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	23.	Internal painted surfaces for scratched, chipped, or peeled paint. Check for any damage to components. Relate scratches and chips to possible points of contact.					
ALL	24.	Harnesses for chafed or broken wires and loose corroded connections. Anti-ice CCA A1 and connector P2 for loose mounting hardware. Window temperature sensor RT1 and thermostat S1 for bonding separation.					
ALL	25.	Electronic shielding gasket for bonding separation, tears, cracks, cuts, and deterioration.					
ALL	26.	PNVS optics for contamination, pitting, nicks, scratching, and peeling of coating.					
ALL	27.	Wire harnesses for chafed or broken wires, loose or corroded connections and loose connectors.					

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspection.

РНА	SE N	0 P	HASE	D MAINTENANCE CHECKLIST			
	ARMA	Area Name and No. AMENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	28. (Check for loose, stripped or missing mounting hardware on PNVS turret.					
ALL		Main fan, mux power supply fan, and two ventilated CCA housings for buildup of dust.					
ALL		Access cover seal for any tears, cracks, cuts, deterioration, and loose fit.					
ALL	31.	Remove PNVS turret assembly.					
ALL		PNVS turret seal for any cracks, tears, bonding separation, or deterioration. Gears for any cracked, broken, or missing teeth. Telfon limit stop pads for any damage.					
ALL		PNVS turret captive mounting screws, stripped, damaged or missing screws.					
ALL		Broken or bent switch actuators and loose limit switches.					

PHA	ASE NO P	HASED MAINTENANCE CHECKLIST						
	Area Name and No. ARMAMENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	3		
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks	and/or Remarks Action Taken		Initial		
ALL	35. Move TADS turret to 90 degrees. Check external painted surfaces of PECA for scratched, chipped, or peeled paint.							
ALL	 Check PECA warning plate decal for peeling edges. Check for extensive wear, pitting, legibility, etc. 							
ALL	 ALQ 136 antenna and associated wiring for mounting security, external damage, and loose connectors. Wiring harnesses and clamps for chafing security, and deterioration. 							

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspection.

PHASE NO PHASED MAINTENANCE CHECKLIST											
Area Name and No. CREW STATIONS – 2				Aircraft Serial No.	Date	Total Hrs. This Area					
Inspect Phase Nos.		Inspection Requirements		Faults and/or Remarks	Action Taken		Initial				
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing attaching hardware. Handholds and steps for damage, structural integrity, and mounting security.									
ALL	2.	Access panels, fairings, and doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges for damage, binding and security. Latches for security and proper operation. Check for worn or deteriorated seals.									
ALL	3.	Interior components for mounting security and loose or missing hardware.									
2,4	4.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.									

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

				a Name and No. W STATIONS – 2			Aircraft Serial No.	Date	
Inspect Phase Nos.		Inspection Requirements			Faults and/or Rema	irks	Action Taken		Initial
2,4	5.	Pilot door for seal deterioration and proper fit. Hinges for cracks and corrosion. Door mechanism for damage and loose door strut cable. Cracked or broken striker plate. Door strut for corrosion and loose or worn bearings.							
2,4	6.	. Remove pilot floor panels for access. Pilot magnetic brake trim and feel spring units for cracks, corrosion, and mounting security. Check for looseness or lost motion. Wiring harnesses for loose connections, chafing, or deteriorations.							
2,4 C	7.	 Pilot flight control linkage for cracks, corrosion, and security. 							
ALL	8.	High powered switching modules for mounting security. Wiring harnesses for proper connections, chafing, and cleanliness. Circuit breakers for looseness and damage. Check all markings for readability.							

РНА	SEI	NO		a Name a N STAT	and No. IONS – 2		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	9.		d for cracks, corrosion, dware. Loose, cracked, nment fittings.						
ALL	10.	Pilot seat for cracks, distortion, and security Upholstery and cushions for tears and cleanliness.							
ALL	11.	 Pilot seat belt and harness straps for cuts, fraying, and cleanliness. Strap fittings for corrosion and security. 							
ALL	12.	Pilot seat belt buckle	for proper operation.						
ALL	13.	 Pilot seat harness reel for strap lock and release operation. Released harness straps for free extension and reel-in operation. 							
ALL	14.	Pilot seat armor swing plate for mounting security, lockup, lock release, and side swing operation.							

РНА	SE NO		a Name an W STATIO			Aircraft Serial No.	Date	
Inspect Phase Nos.	Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	switches for damage	r missing fasteners. ss, control knobs and or looseness. Lenses I cleanliness. Check all						
ALL C	bushings. Grip switcl looseness. Check all readability. Base wiri connections, chafing Control linkage for da	Pilot cyclic stick for security and worn bushings. Grip switches for damage or looseness. Check all markings for readability. Base wiring harness for loose connections, chafing, and deterioration. Control linkage for damage, looseness, and evidence of interference.						
ALL C	bushings. Grip switcl looseness. Check all readability. Base wiri connections, chafing Control linkage for da	Pilot collective stick for security and worn bushings. Grip switches for damage or looseness. Check all markings for readability. Base wiring harness for loose connections, chafing, and deterioration. Control linkage for damage, looseness, and evidence of interference.						
ALL C	security. Supports fo corrosion. Rod and r bending, corrosion, v	B. Pilot directional pedals for damage and security. Supports for cracks, bends, or corrosion. Rod and rod ends for cracks, bending, corrosion, worn or seized bearings, and loose or missing hardware.						

РНА	SE NO			a Name a N STAT	and No. IONS – 2		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4 C	crack brack seize hard	 Pilot engine power controls for bent, cracked or broken cable supports, and brackets. Rods and rod ends for worn or seized bearings and loose or missing hardware. Remove pilot's left console side panels for access. 							
2,4 C	attac conn	 Pilot wiring harnesses for loose and missing attaching hardware, chafed wires, loose connections, and broken tie-wraps. Brackets for damage and corrosion. 							
2,4		SSU mount pad loose or missing	ls for damage, security hardware.						
ALL C	broke	 Pilot canopy jettison components for cut or broken detonation cords, bulged unions, and loose or missing hardware. 							
ALL C	secu	Doghouse fairing for damage. Check for security and integrity of mounted jammer antenna, IFF antenna, and de-ice probe.							

РНА	SEI	NO		a Name a W STAT	and No. ONS – 2		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	24.	CPG seat harness re release operation. Re for free extension and	elease harness straps						
2,4	25.	CPG door for seal deterioration and proper fit. Hinges for cracks or corrosion. Door mechanism for damage and loose door strut cable. Striker plate broken or cracked. Door strut for damage and worn or seized bearings.							
ALL	26.	. CPG seat armor swing plate for mounting security. Check plate lockup for lock release and side swing operation.							
ALL	27.	CPG seat for cracks, distortion, and security. Upholstery and cushions for tears and cleanliness.							
ALL	28.	CPG seat height adju limiter.	ustment track for height						
ALL	29.	. CPG seat belt and harness straps for cuts, fraying, and cleanliness. Strap fittings for corrosion and security.							

РНА	SEI	NO		a Name a N STATI	and No. ONS – 2		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL	30.	consoles for loose or	missing fastners. knobs and switches for s. Lenses for anliness. Check all						
ALL	31.	CPG seat belt buckle	o for proper operation.						
ALL C	32.	 CPG cyclic stick for security and worn bushings. Grip switches for damage or looseness. Check all markings for readability. Base wiring harness for loose connections, chafing, and deterioration. Control linkage for damage, looseness, and evidence of interference. 							
ALL C	33.	. CPG collective stick for security and worn bushings. Grip switches for damage or looseness. Check all markings for readability. Base wiring harness for loose connections, chafing, and deterioration. Control linkage for damage, looseness, and evidence of interference.							

РНА	SE I	NO		a Name a W STATI	and No. ONS – 2		Aircraft Serial No.	Date	
Inspect Phase Nos.		Inspection Requirements			Faults and/or Rema	arks	Action Taken		Initial
ALL C	34.		cracks, bends, and						
ALL C	35.	CPG engine power controls for bent, cracked, or broken cable supports and brackets. Rods and rod ends for worn or seized bearings and loose or missing hardware. Remove CPG left console side panel for access.							
ALL C	36.	 flight control linkage for cracks, corrosion and security. CPG wiring harnesses for loose or missing hardware, chafed wires, loose connections, and broken tie-wraps. Brackets for damage and corrosion. 							
2,4 C	37.								
2,4	38.								

РНА	PHASE NO			Area Name and No. CREW STATIONS – 2			Aircraft Serial No. Date		
Inspect Phase Nos.	Inspection Requirements		Status	Faults and/or Rema	rks	Action Taken		Initial	
ALL C	 CPG canopy jettison components for cut or broken detonation cords, bulged unions, and loose or missing hardware. 								
ALL	cracks, and I	oose or es for lea	rlinders for leakage, missing hardware. akage, chafing, and						
2,4	 Pilot directional pedal control linkage cover for cracks, corrosion, distortion, and loose or missing hardware. (access through gun turret cavity). 								

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
	A	Area Name and No. ARMAMENT CREW STATIONS – 2		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Remove pilot HDU from crewstation.					
ALL	2.	Visually inspect pilot SSU, HDU, and BRU optical lenses for contamination, clean as required.					
ALL	3.	Inspect pilot BRU housing for damage. None allowed.					
2,4	4.	Inspect pilot BRU wiring harness, connector, and receptacle for loose, broken, or missing pins or damage. None allowed.					
ALL	5.	Inspect pilot display adjust panel (DAP) for cracks, corrosion, damaged connector receptables, and loose or missing hardware. Cracks: None allowed. If found, replace with undamaged like item.					

РНА	SE	NO		a Name a T CREW	and No. / STATIONS – 2		Aircraft Serial No.		e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL	6.	Pilot Sensor Surveyir components for cracl connector receptacle allowed, if found repl like item.	ks, and damaged . Damage: none						
ALL	7.	Visually inspect CPG SSU, HDU, and BRU optical lenses for contamination, clean as required.							
ALL	8.	. Remove CPG HDU from crewstation.							
ALL	9.	Inspect CPG BRU ho None allowed.	ousing for damage.						
ALL	10.	 Inspect CPG display adjust panel (DAP) for cracks, corrosion, damaged connector receptacles and loose or missing hardware. Cracks; none allowed. If found, replace with undamaged like item. 							
ALL	11.	 CPG Sensor Surveying Unit (SSU) component for cracks, and damaged connector receptacle. 							

РНА	SE NO		a Name T CREW	and No. / STATIONS – 2		Aircraft Serial No.	Date	
Inspect Phase Nos.	Inspection Re	equirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	12. Remove combiner le attaching parts.	ens from HDU and clean						
ALL	 Inspect the HDU ho spring, and thrust pl replacement damag 							
ALL		broken, bent, loose or missing pins and						
ALL	15. Check combiner len scratches and for lo guard.	s for chips, cracks, and ose, broken or missing						
ALL	16. Inspect CRT cable f	or twisting or binding.						
ALL	17. HDU for damage, cr or missing hardware	racks, corrosion, loose e, or connectors.						
ALL	and two screws for	 Rotating ring, rotating segment, flat spring, and two screws for negligible or replacement damage. 						

РНА	PHASE NO		Area Name and No. ARMAMENT CREW STATIONS – 2			Aircraft Serial No.			e
Inspect Phase Nos.	Inspe	Inspection Requirements			Faults and/or Rema	irks	Action Taken		Initial
ALL	19. Perform HDL	Perform HDU alignment verification.							
ALL	or peeling pa	or peeling paint, optics for contamination, pitting, nicks, scratching and ORT humidity							
ALL		Control panel filter assembly for unrestricted movement and security in the up position.							
ALL	22. Check for de pads.	0							

РНА	SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
		Area Name and No. LEFT EFAB AND MLG – 3		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose					
		or missing hardware. Handholds and steps for damage, structural integrity, and mounting security.					
ALL	2.	Access panels, fairings, and doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges for damage, binding,					
		and security. Latches for security and proper operation. Check for worn or					
		deteriorated seals and copper cote.					
ALL	3.	Interior structure for cracks, corrosion, loose					
		or working rivets, and loose or missing hardware.					
ALL	4.	ECS compressor, condenser, evaporator					
		and tubes for leaking and integrity. All ECS					
		system components for proper charge, signs of leakage or overheating and proper					
		installation.					
		installation.					

РНА	SE	NO		a Name a FAB AN	and No. ID MLG – 3		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL	5.	Mounting racks for distortion and loose or missing hardware. Cooling ducts for cracks, distortion, proper fit, and evidence of leakage.							
ALL	6.	Avionics bay for cleanliness and distortion.							
ALL C	7.	gear shock strut mou mount to the end of t pits, grooves and scr	atches. For cracks use t. Check for distortion, or missing rivets or						
ALL	8.	Inspect main landing bearings for damage							
2,4 C	9.	Main landing gear tra distortion, and corros pivot boss for securit bearing, loose or mis Hydraulic brake line f dents, corrosion, and	sion. Cross tube end y, damaged or worn ssing hardware. for leakage, chafing,						

РНА	SE NO		ea Name a EFAB ANE	nd No. D MLG – 3		Aircraft Serial No.	Dat	e
Inspect Phase Nos.	Inspectio	n Requirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	cracks, distortior	 Main landing gear end cap on cross tube for cracks, distortion, fractured weld, and elongated bolt holes. 						
2,4 C	11. Repack main lar	1. Repack main landing gear wheel bearings.						
2,4 C		 Main landing gear wheel keys and key slots for wear and damage. 						
ALL	insulation, loose	deformation, damaged connections, and mounting narness for chafing, erioration.						
ALL	interference with	ellcrank for chafing and forward fuel cell. Check for pration and security.						

РНА	SE NO	Area Name and No. LEFT EFAB AND MLG – 3				Aircraft Serial No.	rcraft Serial No. Date		
Inspect Phase Nos.	Inspection Requirements		Status	Faults and/or Rema	rks	Action Taken		Initial	
ALL	15. Pilot collective push-pull rods and rod ends for cranks, corrosion, bending, worn and seized bearings, loose or missing hardware, and evidence of interference.								

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
	AR	Area Name and No. MAMENT LEFT EFAB AND MLG – 3		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Connectors J1 through J9 for corrosion,					
		damage, distortions, and loose mounting connections.					
ALL	2	Ctow cocombly mounting points for creaks					
ALL	2.	Stow assembly mounting points for cracks and corrosion.					
ALL	.L 3.	TEU assembly for loose, missing, or stripped mounting hardware. Remove TEU.					
ALL	4	Air balas in sinflow adaptors for build up of					
ALL	4.	Air holes in air flow adapters for build-up of dust.					
	_	Demous TDO, Oheels TDO (on lease					
ALL	5.	Remove TPS. Check TPS for loose, missing, and stripped mounting hardware.					
		5,					
ALL	6.						
		connections and loose, bent, or broken pins and sockets					
		and sockets.					

PHA	SE	NO		a Name a _EFT EF	and No. AB AND MLG – 3		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	7.	Fuses F1, F2, and F3 for missing, blown, or incorrect fuses.							
ALL	8.	Check air holes in air flow adapters on rear of TPS for build-up of dust.							
ALL	9.	Connectors J2 and J3 for corrosion, loose connections and for loose, bent, or broken pins and sockets.							
ALL	10.	LEU for loose, missir mounting hardware.							
ALL	11.	 Avionics equipment for mounting security, external damage, and loose connectors. Wiring harnesses and clamps for chafing, security, and deterioration. 							
ALL	12.	Wire harnesses for chafing, broken wires, and loose, bent, broken pins and sockets.							

РНА	SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
	_	Area Name and No. RIGHT EFAB AND MLG – 4		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware. Step for damage, structural integrity, and mounting security.					
ALL	2.	Access panels, fairings, and doors for deformation cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges for damage, binding, and security. Latches for security and proper operation. Seals and copper cote for wear or deterioration.					
ALL	3.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	4.	ECS compressor, condenser, evaporators, and tubes for leaking and integrity. All ECS system components for proper charge. Signs of leakage or over heating and proper installation.					

РНА	SE	NO		a Name a EFAB AN	and No. ND MLG – 4		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	5.	Mounting rack for distortion and loose or missing hardware. Cooling ducts for cracks, distortion, proper fit, and evidence of leakage.							
ALL	6.	Avionics bay for cleanliness and distortion.							
ALL C	7.	gear shock strut mou mount to the end of the pits, grooves and scr	atches. For cracks use t. Check for distortion,						
ALL	8.	Inspect main landing bearings for damage							
2,4 C	9.	Main landing gear tra distortion, and corros pivot boss for security bearing, loose or mis Hydraulic brake line f dents, corrosion, and	sion. Cross tube end y, damaged or worn sing hardware. for leakage, chafing,						

РНА	SE NO		a Name a EFAB AN	ind No. ID MLG – 4		Aircraft Serial No.	Date	e
Inspect Phase Nos.	Inspection Red	quirements	Status	Faults and/or Remai	rks	Action Taken		Initial
ALL C	 Main landing gear end cap on cross tube for cracks, distortion, fractured weld, and elongated bolt holes. 							
2,4 C	11. Repack main landing	1. Repack main landing gear wheel bearings.						
2,4 C	12. Main landing gear wh for damage and wea	neel keys and key slots r.						
ALL	 Extend searchlight. Check for corrosion, loose or missing hardware, and security. Lens for cracks and evidence of overheating. Wiring for loose connections, chafing, deterioration, and security. 							

SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
ARI	Area Name and No. MAMENT RIGHT EFAB AND MLG – 4		Aircraft Serial No.	Date	Total Hrs. This Area	
	Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial
1.	SEU and connectors for corrosion, damage, loose mounting and connections.Remove SEU					
2.	. SEU to Electronic Equipment Test Facility (EETF) for testing.					
3.	DEU and connectors for corrosion, damage, loose mounting and connections. Remove DEU.					
4.	DEU to Electronic Equipment Test Facility (EETF) for testing.					
5.	Inspect air hole in air flow adapters on rear of PEU for build-up of dust. Inspect PEU for loose, missing or stripped mounting hardware. Remove PEU.					
	AR 1. 2. 3. 4.	Area Name and No. ARMAMENT RIGHT EFAB AND MLG – 4 Inspection Requirements 1. SEU and connectors for corrosion, damage, loose mounting and connections.Remove SEU 2. SEU to Electronic Equipment Test Facility (EETF) for testing. 3. DEU and connectors for corrosion, damage, loose mounting and connections. Remove DEU. 4. DEU to Electronic Equipment Test Facility (EETF) for testing. 5. Inspect air hole in air flow adapters on rear of PEU for build-up of dust. Inspect PEU for loose, missing or stripped mounting	Area Name and No. ARMAMENT RIGHT EFAB AND MLG – 4 Inspection Requirements Status 1. SEU and connectors for corrosion, damage, loose mounting and connections.Remove SEU	Area Name and No. Aircraft Serial No. ARMAMENT RIGHT EFAB AND MLG – 4 Inspection Requirements Status Faults and/or Remarks 1. SEU and connectors for corrosion, damage, loose mounting and connections.Remove SEU	Area Name and No. Aircraft Serial No. Date ARMAMENT RIGHT EFAB AND MLG – 4 Inspection Requirements Status Faults and/or Remarks	Area Name and No. Aircraft Serial No. Date Total Hrs. This Area ARMAMENT RIGHT EFAB AND MLG – 4 Inspection Requirements Status Faults and/or Remarks Action Taken 1. SEU and connectors for corrosion, damage, loose mounting and connections.Remove SEU

РНА	SEI	NO		a Name a IGHT EI	and No. FAB AND MLG – 4		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
ALL	6.	Remove fuseholder caps for F1 through F4. Check for missing, blown, or incorrect fuses.							
ALL	7.	Connectors J1 through J5 for corrosion and loose connections. Check for loose, bent, burned, or broken pins and sockets.							
ALL	8.	Avioncs equipment for mounting security, external damage, and loose connectors. Wiring harnesses and clamps for chafing, security and deterioration.							
ALL	9.	Wire harnesses for c loose, bent, broken p							
ALL	10.	0. Gun control box (GCB) and connectors for corrosion, damage, loose mounting and connectors.							
ALL	11.	 Turret control box (TCB) and connectors for corrosion, damage, loose mounting, and connections. 							

РНА	SE NO		Area Name and No. NT RIGHT EFAB AND MLG – 4			Aircraft Serial No.	Dat	e
Inspect Phase Nos.	Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	security, proper oper	 Sideloader magazine controller (SMC) for security, proper operation, broken or bent gears, shaft carriers and slide tray. 						

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
	_	Area Name and No. LEFT WING AND PYLON – 5		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	2.	Access panels, fairings, and wing tip door for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinge and latch for damage, binding, security, and proper operation. Seals for wear and deterioration.					
ALL	3.	Inspect wing forward upper mounting flange area and mounting bolts for security, cracks and corrosion.					
ALL	4.	. Interior components for mounting security and loose or missing hardware.					
ALL	5.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.					

РНА	SE	NO		a Name NG AND	and No. PYLONS – 5		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	6.	Wing mount fittings for and distortion.	/ing mount fittings for cracks, corrosion nd distortion.						
ALL	7.	security, chafing, and	utboard pylon wiring harnesses for ty, chafing, and loose connections. ulic and fuel lines for security, chafing, akage.						
ALL	8.	security, chafing, and	eft inboard pylon wiring harnesses for ecurity, chafing, and loose connections. lydraulic and fuel lines for security, chafing, nd leakage.						
ALL	9.	Wing wiring harnesse and proper connection	es for security, chafing, on.						
2,4	10.	working rivets. Hydra chafing, and clampin	for cracks, corrosion, and loose or g rivets. Hydraulic lines for leakage, g, and clamping security. Wiring sees for chafing and clamping y.						
2,4	11.	 Pitot/static lines for cracks, chafing, and mounting security. 							

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
	ARM	Area Name and No. IAMENT LEFT WING AND PYLON – 5		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL C	1.	cracks, dents, distortion, and loose or missing fastners. Pylon and rack mount fittings for cracks and distortion. Mounting bolts for security.					
ALL	2.	Pylon fairings and actuators for evidence of damage and hydraulic fluid leakage.					
ALL	3.	Access P3. Pylon PIU bracket for cracks, corrosion, or warping.					
ALL	4.	Check pylon connectors for corrosion, damage, and loose mounting connections.					
ALL	5.	Check pylon rack inserts, bushings, and bearings for visible damage.					
ALL	6.	Check pylon rack mounting bushings.					
ALL	7.	Check pylon hollow pin.					

РНА	SEI	NO		a Name a FT WIN	and No. G AND PYLONS – 5		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	8.	Check pylon frame assembly for cracks on both inboard and outboard sides. Use magnifier around pivot pin area. No cracks allowed.							
ALL C	9.	Pylon ejector assembly for corroded piston and cartridge holder.							
ALL	10.	Check pylon rack art	iculation.						
ALL	11.	Check pylon rack for torque.	allowable release						
ALL C	12.	Left inboard installed cracks, dents, distort missing fastners. Pyl fittings for cracks and bolts for security.	ion, and loose or on and rack mount						
ALL	13.	Pylon fairings and ac damage and hydraul	ctuators for evidence of ic fluid leakage.						
ALL	14.	Access P3. Pylon PII corrosion, or warping							

РНА	SEN	NO		a Name a FT WIN(and No. G AND PYLONS – 5		Aircraft Serial No.	Dat	e
Inspect Phase Nos.	Inspection Requirements			Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	15.	 Check pylon connectors for corrosion, damage, and loose mounting connections. 							
ALL	16.	. Check pylon rack inserts, bushings, and bearings for visible damage.							
ALL	17.	7. Check pylon rack mounting bushings.							
ALL	18.	3. Check pylon hollow pin.							
ALL	19.	both inboard and outboard sides. Use magnifier around pivot pin area. No cracks allowed.							
ALL C	20.								
ALL	21.	. Check pylon rack articulation.							
ALL	22.	 Check pylon rack for allowable release torque. 							

РНА	SE NO		a Name a FT WIN(and No. G AND PYLONS – 5	Aircraft Serial No.		Date	
Inspect Phase Nos.	Inspection Requirements		Status	Faults and/or Rema	irks	Action Taken		Initial
ALL	23. Perform missile laun	23. Perform missile launcher inspection.						
ALL		 Rocket pod for wear, cracks, deterioration, bends, and dents in tubes. 						
ALL	25. Rocket pod connecto pins.	•						
ALL	26. Igniter arms for prop	6. Igniter arms for proper operation.						
ALL	7. Lubricate rocket pods.							

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
	_	Area Name and No. RIGHT WING AND PYLONS – 6		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	2.	Access panels, fairings, and wing tip door for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinge and latch for damage, binding, security, and proper operation. Seals for wear or deterioration.					
ALL	3.	Inspect wing forward upper mounting flange area and mounting bolts for security, cracks and corrosion.					
ALL	4.	Interior components for mounting security and loose or missing hardware.					
ALL	L 5. Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.						

РНА	SEI	NO		a Name a NG ANE	and No.) PYLONS – 6		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL C	6.	Wing mount fittings for cracks, distortion and corrosion.							
ALL	7.	security, chafing, and	ight outboard pylon wiring harnesses for ecurity, chafing, and loose connections. lydraulic and fuel lines for security, chafing nd leakage.						
ALL	8.	security, chafing, and	ing harnesses for security, chafing,						
ALL	9.	Wing wiring harnesse and proper connection							
2,4	10.	Spars for cracks, corrosion, and loose or working rivets. Hydraulic lines for leakage, chafing, and clamping security. Wiring harnesses for chafing and clamping security.							

РНА	SE NO	Area Name and No. RIGHT WING AND PYLONS – 6				Dat	e	
Inspect Phase Nos.	Inspection Requirements		Status	Faults and/or Rema	rks	Action Taken		Initial
2,4	11. Pitot/static lines for c mounting security.	 Pitot/static lines for cracks, chafing, and mounting security. 						
С								

РНА	SE	NO P	PHASE	MAINTENANCE CHECKLIST			
A	RMA	Area Name and No. MENT RIGHT WING AND PYLONS – 6		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL C	1.	Right outboard installed pylons and racks for cracks, dents, distortion, and loose or missing fastners. Pylons and rack mount fittings for cracks and distortion. Mounting bolts for security.					
ALL	 Pylon fairings and actuators for evidence o damage and hydraulic fluid leakage. 						
ALL	3.	Access P3. Pylon PIU bracket for cracks, corrosion, and warping.					
ALL	4.	Check pylon connectors for corrosion, damage, and loose mounting connections.					
ALL	5.	Check pylon rack inserts, bushings, and bearings for visible damage.					
ALL	6.	Check pylon rack mounting bushings.					
ALL	L 7. Check pylon hollow pin.						

РНА	SE I	NO		a Name a HT WIN	and No. IG AND PYLONS – 6		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	8.	Check pylon frame assembly for cracks on both inboard and outboard sides. Use magnifier around pivot pin area.							
ALL C	9.	Pylon ejector assembly for corroded piston and cartridge holder.							
ALL	10.	Check pylon rack articulation.							
ALL	11.	Check pylon rack for torque.	eck pylon rack for allowable release que.						
ALL C	12.	cracks, dents, distort missing fastners. Pyl	ht inboard installed pylons and racks for cks, dents, distortion, and loose or sing fastners. Pylon and rack mount ngs for cracks and distortion. Mounting as for security.						
ALL	13.	Pylon fairings and ac damage and hydraul	ctuators for evidence of ic fluid leakage.						
ALL	14.	Access P3. Pylon PII corrosion, or warping	U bracket for cracks, J.						

PHA	SEI	NO		a Name a HT WIN	and No. G AND PYLONS – 6		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	15.	Check pylon connectors for corrosion, damage, and loose mounting connections.							
ALL	16.	Check pylon rack inserts, bushings, and bearings for visible damage.							
ALL	17.	. Check pylon rack mounting bushings.							
ALL	18.	Check pylon hollow pin.							
ALL	19.	Check pylon frame as both inboard and out magnifier around pive							
ALL C	20.	Pylon ejector assemb and cartridge holder.	bly for corroded piston						
ALL	21.	Check pylon rack articulation.							
ALL	22.	Check pylon rack for allowable release torque.							

РНА	PHASE NO ARMAMENT		a Name GHT WIN	and No. IG AND PYLONS – 6		Aircraft Serial No.	Dat	e
Inspect Phase Nos.	Inspection Requirements		Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	23. Perform missile launcher inspection.							
ALL		24. Rocket pod for wear, cracks, deterioration, bends, and dents in tubes.						
ALL	25. Rocket pod connect pins.	•						
ALL	26. Igniter arms for prop	per operation.						
ALL	7. Lubricate rocket pods.							

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
	LEF	Area Name and No. T ENGINE AND NOSE GEARBOX – 7		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	2.	Access panels, fairings, and doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges, struts, and supports for damage, binding, and security. latches for security and proper operation. Seals for wear and deterioration.					
ALL C	3.	Navigation, and anti-collision lights for corrosion, loose or missing hardware, and security. lenses for cracks, looseness, and discoloration. Wiring harness for chafing and clamping security.					
ALL	4.	IR suppressor nozzle mating surfaces for worn, and damaged orange seals.					
ALL	5.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.					

РНА	SEI	NO		a Name a AND NC	and No. DSE GEARBOX – 7		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4 C	6.	Particle separator duct for cracks, dents, and deformation.							
2,4	7.	IR suppressor nozzles and radiation shields for cracks, dents, deformation, and security. Supports for damage and security.							
2,4	8.	Main transmission heat exchangers and engine cooloing louvers for cracks, deformation, delamination, distortion, broken and loose or working rivets. Louver plates for looseness and lost motion.							
ALL C	9.	. Engine air inlet for cracks, distortion, security, corrosion, deteriorated, torn and split seals. Loose or working rivets and screws. Interior for cleanliness.							
2,4	10.	Drain and service en cracks and loose or r	gine starter. Check for nissing hardware.						
2,4	11.	 Engine wiring harnesses for loose connections, chafing, and deterioration. 							

РНА	SE NO			a Name a AND NC	and No. DSE GEARBOX – 7		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4 C	for crac	for cracks, corrosion, mounting security, and loose or missing hardware.							
ALL C	13. Primary	3. Primary exhaust nozzle for cracks.							
2,4 C	deforma security	ation, loose by , and loose o d expanding l	ne mounts for cracks, ushings, corrosion, r missing hardware. bolts for wear and						
2,4	15. Nose go chafing	earbox wiring , and deterior	for loose connections, ation.						
ALL C			ose gearbox chip Isulation damage.						
2,4 C	17. Remove	e and clean n	ose gearbox breather.						
ALL C	18. Change	e nose gearbo	ox lube , oil, and filter.						

РНА	SEN	NO		a Name a AND NOS	nd No. SE GEARBOX – 7		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4	19.	 Nose gearbox oil pressure switch, pressure transducer, and temperature probe for insulation damage, oil leakage, and security. Harness splices for security. 							
2,4	20.	 Nose gearbox lube oil level sight gage for cleanliness, leakage, and security. 							
ALL C	21.	 Nose gearbox and lube oil pump housings for cracks, distortion, leakage, and security. 							
2,4 C	22.	Nose gearbox mount is 60 inch/lbs.	ting bolts, verify torque						
2,4 C	23.	 Nose gearbox cooling fins and shroud for cracks, deformation, corrosion, and security. Fan impeller for corrosion and damage. 							
ALL C	24.	Nose gearbox drive s nicks, dents, scratche	shaft and couplings for es, and security.						

PHA	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
	RIGH	Area Name and No. IT ENGINE AND NOSE GEARBOX – 8		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	2.	Access panels, fairings, and doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges, struts, and supports for damage, binding, and security. Latches for security and proper operation. Seals for wear or deterioration.					
ALL C	3.	Navigation and anti-collision lights for corrosion, loose or missing hardware, and security. Lenses for cracks, looseness, and discoloration. Wiring harness for chafing and clamping security.					
ALL	 IR suppressor nozzle mating surface for worn and damaged orange seals. 						

РНА	SE	NO		a Name an AND NO	n d No. SE GEARBOX – 8		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	5.	Interior structure for cracks, corrosion. loose or working rivets, and loose or missing hardware.							
2,4 C	6.	Particle separator duct for cracks, dents, and deformation.							
2,4	7.	IR suppressor nozzles and radiation shields for cracks, dents, deformation, and security. Supports for damage and security.							
2,4	8.	Main transmission heat exchanger and engine cooling louvers for cracks, deformation, delamination, distortion, broken or loose rivets. Louver plates for looseness and lost motion.							
ALL C	9.	Engine air inlet for cracks, distortion, security, corrosion, deteriorated, torn and split seals. Loose or working rivets and screws. Interior for cleanliness.							

РНА	SE NO	-	a Name an AND NOS	nd No. SE GEARBOX – 8		Aircraft Serial No.	Date	e
Inspect Phase Nos.	Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4	10. Drain and service en cracks and loose or r	gine starter. Check for nissing hardware.						
2,4		Engine wiring harnesses for loose connections, chafing, and deterioration.						
2,4 C	for cracks, corrosion,	 Engine power control cables and brackets for cracks, corrosion, mounting, security, and loose or missing hardware. 						
ALL C	13. Primary exhaust noz	zle for cracks.						
2,4 C	deformation, loose be security, and loose of	deformation, loose bushings, corrosion, security, and loose or missing hardware. Pins and expanding bolts for wear and						
2,4		 Nose gearbox wiring for loose connections, chafing, and deterioration. 						

РНА	SEI	NO		a Name AND N	and No. OSE GEARBOX – 8		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
ALL C	16.	Remove and clean n detector. Check for ir							
2,4 C	17.	7. Remove and clean nose gearbox breather.							
ALL C	18.	8. Change nose gearbox lube, oil and filter.							
2,4	19.	 Nose gearbox oil pressure switch, pressure transducer, and temperature probe for insulation damage, oil leakage, and security. Harness splices for security. 							
2,4	20.	 Nose gearbox lube oil level sight gage for cleanliness, leakage, and security. 							
ALL C	21.	Nose gearbox and lu for cracks, distortion,	be oil pump housing leakage, and security.						
2,4 C	22.	Nose gearbox mount is 60 inch/lbs.	ing bolts, verify torque						

PHA	PHASE NO		Area Name and No. RIGHT ENGINE AND NOSE GEARBOX – 8			Aircraft Serial No. Da			e
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	rks	Action Taken		Initial
2,4 C	23.	 Nose gearbox cooling fins and shroud for cracks, deformation, corrosion, and security. Fan impeller for corrosion and damage. 							
ALL C	24.	 Nose gearbox drive shaft and couplings for nicks, dents, scratches, and security. 							

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
		Area Name and No. MAIN TRANSMISSION – 9		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	2.	Access panels, fairings, and doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges for damage, binding, and security. Latches for security and proper operation. Seals and copper coat for wear or deterioration.					
ALL	3.	Interior components for mounting security and loose or missing hardware.					
ALL	4.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	5.	Transmission housing and cover for cracks, oil leakage, and evidence of overheating (discoloration).					

РНА	SEI	NO		a Name RANSM	and No. ISSION – 9		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	6.	Torque check upper	case nuts to 190 in. lbs.						
ALL C	7.	7. Remove and clean chip detectors. Check for insulation and damage.							
2,4	8.	 Wiring harnesses for loose connections, chafing, or deterioration. 							
ALL C	9.	. Replace accessory pump oil filter. Remove and clean bypass screen.							
2,4 C	10.	Clean transmission b	preathers.						
ALL C	11. Change transmission lube oil and filters.								
2,4	12. Oil pressure switches, pressure transducers, temperature probes, and magnetic pickup for insulation damage, leakage, and security. Harness splices for security.								

РНА	SE	NO		a Name RANSM	and No. IISSION – 9		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4	13.	Reaction bar for crac	ks and corrosion.						
ALL	14.	14. Reaction bar for signs of wear, presence of metal fragments, elongated hole, and torque striping on two clamping bolts.							
ALL	15.	 Lube oil level sight gages for cleanliness, leakage, and security. 							
2,4 C	16.	 Input shaft and coupling bolts for proper installation. Verify torque on bolts. 							
ALL C	17.	 Primary hydraulic manifold for leakage, corrosion, loose connections, and security. Check manifold air inlet check valve filter for cleanliness. 							
2,4 C	18.	 Generators for damaged insulation, security, and cracked or broken housings. 							

	SE NO		a Name RANSM	and No. ISSION – 9		Aircraft Serial No.	Dat	e
Inspect Phase Nos.	Inspection Rec	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
2,4	19. Transformer/rectifiers overheating (discolor							
ALL C	20. Anti-collision light por corrosion and loose c	wer supply for or missing hardware.						
ALL C	cracks, and corrosion ends for worn and se boot for cuts, tears, a Inspect servo cylinde	Flight control servo cylinders for leakage, cracks, and corrosion. Upper and lower rod ends for worn and seized bearings. Rubber boot for cuts, tears, and deterioration. Inspect servo cylinder control linkage hardware for damage and security.						
ALL C	bending, distortion, a Transmission deck fo	 Main rotor mast support struts for cracks, bending, distortion, and security. Transmission deck for distortion and looseness at lower ends of struts. Mast base for cracks, distortion, and security. Mast support mount and upper ends of support struts for security. Inspect the upper portion of the mast base support in the areas around the four lighting holes, static mast, and mixer supports for corrosion. 						
ALL C	security. Mast suppor ends of support struts the upper portion of t in the areas around t static mast, and mixe							

РНА	SE NO		a Name and N RANSMISSI			Aircraft Serial No.	Dat	e
Inspect Phase Nos.	Inspection Re	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL C	24. Static mast attaching torque (use PLI was	g hardware for proper her method).						
ALL C	cracks, corrosion, se	Flight control rod and rod ends for dents, cracks, corrosion, security, and worn or seized bearings and bushings.						
2,4 C	corrosion, and secur	corrosion, and security. Brackets for mounting security. Pivot bearings for						
2,4 C	27. Engine controls for damage and deformed cables, supports, clamps, and brackets for cracks and bends. Rods and rod ends for worn and seized bearings. Bellcranks for cracks, deformation, worn bushings, loose or missing hardware, and evidence of interference.							

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
	_	Area Name and No. MAIN ROTOR – 10		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
2,4 C	1.	Blade spars and root finger doublers for delamination.					
2,4 C	2.	Blade leading edge for cracks, dents, distortion, and erosion. Leading edge tips for loose or missing hardware. Trailing edge for cracks and voids.					
ALL C	3.	Perform coin tap test on main rotor blades.					
2,4 C	4.	Blade root bushings for cracks, distortion, and security.					
2,4 C	5.	. Main rotor drive plate for cracks, distortion, and corrosion. Mounting bolts for security.					

РНА	SE	NO	-	a Name a N ROTC			Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4 C	6.	Main rotor hub for cra corrosion, and securi seal for leakage.							
2,4 C	7.	Feathering bearing h nuts for cracks, corro Feathering bearings							
2,4 C	8.	Blade droop stop striker plates for distortion and cracks. Droop stop plungers, return springs, and stop ring for cracks and deformation. Plungers, rollers, and stop ring for wear and play.							
2,4 C	9.	Pitch housings for cra corrosion, and securi for pitch link rod end	ity. Pitch housing ears						
2,4 C	10.	corrosion. Link-to-damper rod ends for bearing damage. Rod ends for worn and seized bearings.							
2,4 C	11.								

РНА	SE NO	-	a Name and IN ROTOR -			Aircraft Serial No.	Dat	e
Inspect Phase Nos.	Inspection Re	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
ALL C	breakage or horizon	 Blade strap packs for cracks, buckling, breakage or horizontal displacement. (Check both ends of each strap). 						
2,4 C	distortion, cracked a	Static discharge brushes for fraying, distortion, cracked and broken holders or springs. Brushes for even contact with mast.						
ALL C	corrosion. Rod ends	corrosion. Rod ends for bearing damage or looseness. Check lower rod end clamp-up						
ALL C		flaked or worn-through plating. Bearing for						
ALL C	grease leakage, and Pitch link connection	 Rotating swashplate for cracks, corrosion, grease leakage, and security of lower seal. Pitch link connection bosses for bending, misalignment, and worn or loose bushings. 						

РНА	SE NO		a Name a N ROTO			Aircraft Serial No.	Dat	e
Inspect Phase Nos.	Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	 Stationary swashplat corrosion. Lateral and bosses for bending, r or loose bushings. 							
ALL C		otating scissors for cracks, corrosion, and ecurity. Pivot bearings for wear.						
ALL C	cracks, dents, scratc	cracks, dents, scratches, and corrosion. Attachment bolts for security. Bearings for						
ALL C	bellcracks for cracks	bellcracks for cracks, distortion, and corrosion. Check floating bushing clamp-up						
2,4 C	corrosion. Mixer attac cracks, corrosion, an	Mixer supports for cracks, distortion, and corrosion. Mixer attachment bolts for cracks, corrosion, and security. Check for worn or seized bearings.						

РНА	SEI	NO		a Name a N ROT(Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	22.	Lower shoes for worr worn plunger bearing	n scissor bearings and ls.						
ALL C	23.								
ALL C	24.	Check rotor hub static droop angle.							
2,4 C	25.	Static mast for cracks, dents, distortion, and corrosion. Swashplate sliding surface for grooved, flaked, and worn-through plating.							
ALL C	26.	Derotation unit for da deformation (If install	mage, looseness, and ed).						
ALL C	27.	Mast base flange for cracks, corrosion, and security.							

РНА				and No. DR – 10	Aircraft Serial No.		Date	
Inspect Phase Nos.	Inspection F	Inspection Requirements		Faults and/or Rema	arks	Action Taken		Initial
2,4 C	magnifying glass. I	. Check main rotor upper bearing using 12X magnifying glass. Inspect bearing for burnt grease and contamination. Repack bearing.						
ALL C	corrosion, and for	. Main rotor gear shaft for cracks, distortion, corrosion, and for chipped, broken, scored or worn spline teeth.						

РНА	SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
		Area Name and No. ARMAMENT MAIN ROTOR – 10		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL C	1.	 Remove the 12 MMA to DU attachment bolts and separate the MMA from the DU. Visually inspect at 10X magnification minimum, the DU attachment bolt holes (12) and nut plate cage rivet holes (24) for cracks on the bottom surface of the stationary pedestal. Also visually inspect the inside surface of the stationary pedestal for cracks (opposite the lower azimuth bearing journals). 					
ALL C	2.	Verify integrity of torque stripes on baseplate attachment to pedestal shelf (12 screws external).					
ALL C	3.	Unfasten the 26 Radome/Aft Dome bolts and remove the Radome.					
ALL C	4.	Unfasten the 25 Aft Dome/Baseplate bolts and open the Aft Dome for internal inspection.					

PHA	SE	NO	-	a Name an NT MAIN F	d No. ROTOR – 10		Aircraft Serial No.		e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	5.	torque stripes for the Twenty-two Hub Col Twenty-four Hub Col Any fastner with torq stripe chafed, misalig should be inspected Visually inspect all ur	tached wave guides ove the Line (LRM) from the o inspect and verify the following: lar/Rotary Tube Bolts; llar/Baseplate Bolts. ue stripe damage (i.e. gned, or missing) and fully torqued. hobited and unfastened and around bolt/fastner						

PH/	ASE NO.		-	ea Name and N NT MAIN RO			Aircraft Serial No.		e
Inspect Phase Nos.	Ins	pection Re	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL C	following ex torque strip re-torque a Four Azimu Screws – to Six Azimut Screws. Four RFI A Four RFI R (on units w Two Time N	xternal bo be. None a and apply uth Drive A wo places h Drive El ntenna Ca eceiver C eceiver B here insta Meter Scre Particle S	ectronic Unit Captive aptive Screws. aptive Screws. racket Captive Screws Illed). ews.						

PHA	SEI	NO		a Name a	and No. I ROTOR – 10		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL C	7.	cracks, or grazing at of the top flat surface bolts, eight inches for inches wide along the the top flat surface ar radius, four inches ei line. Any noticable fla inch diameter should AMCOM Engineering monitored. A flaw siz inch requires repair b	earing as delamination, the bolting flange; area a from the captive stud rward; and area eight e tangent line between and the upper corner ther side of the tangent aw sized less than 1/2 be reported to g and subsequently ed greater than 1/2						
ALL C	8.	Install Transmitter.							
ALL C	9.	Close AFT dome.							
ALL C	10.	Install Radome.	nstall Radome.						

PHA	SE NO		a Name and I NT MAIN RO		Air	rcraft Serial No.	Date	9
Inspect Phase Nos.	Inspection Rec	quirements	Status	Faults and/or Remar	ks	Action Taken		Initial
ALL C	for minimum run in to Torque shall be appli bolts diametrically op (at right angle) to the followed by pairs of b pairs. The first stage the bolt against the c The second and third the bolts to 50 percent	ed in stages to pairs of posed at 90 degrees first pair. This shall be polts between first two shall snug the head of ontacting structure.						
ALL C	following bolts and so 12 MMA to DU attack	following bolts and screws: 12 MMA to DU attachment bolts; 12 screws on baseplate attachment to						

PHA	SE	NO		a Name a NT MAIN	nd No. ROTOR – 10	Aircraft Serial No.		Date	e
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	13.	stripes to following ex Four azimuth actuato places); Six azimuth amplifier Four RFI antenna cap Four RFI receiver cap Four RFI receiver bra Two time meter screw Ten inertial particle se	ARMAME Inspection Requirements heck physical security and apply torque ipes to following external bolts/screws: ur azimuth actuator captive screws (2						

РНА	SE	NO P		IAINTENANCE CHECKLIST			
		Area Name and No. AFT EQUIPMENT BAY – 11		Aircraft Serial No.	Date	Total Hrs. This Are	ea
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	A	ction Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware. Handholds for damage, structural integrity, and mounting security.					
ALL	2.	Access doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges for damage, binding, and security. Latches and lanyards for security and proper operation. Seals for wear and deterioration.					
ALL C	3.	Structural mating surfaces and fairings for worn or deteriorated seals and copper coat.					
2,4 C	4.	Tail rotor drive shaft and coupling bolts for proper installation. Verify torque on bolts.					

РНА	SE	NO		a Name a UIPMEN	and No. JT BAY – 11		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4 C	5.	bearings for cracks, or security, and for radia smooth operation. Ha smooth rotation. Veri	al looseness and anger bearings for						
2,4 C	6.	Main transmission heat exchangers and louver actuators for cracks, deformation and security.							
2,4 C	7.	security. Cartridges, t cracks, distortion and extinguisher charge p	Fire extinguisher for dents and mounting ecurity. Cartridges, fittings and valves for racks, distortion and security. Check fire extinguisher charge pressure. Inspect lischarge–indicating disk.						
2,4 C	8.	Fire extinguisher outl for cracks, dents, nic distortion, and securi tubes around "B" nut	ty. Inspect interior of						
2,4 C	9.	Fire extinguisher system check valves for interior corrosion, pitting, and evidence of evaporation							

РНА	SEI	NO		a Name a UIPMEN	and No. IT BAY – 11		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	10.	APU enclosure cover cracks. delamination missing fastners, and APU closure, chafing and deck.	, seal damage, loose or d security. Check for						
ALL	11.	APU drive shaft and coupling for cracks, dents, distortion, corrosion, and evidence of interference.							
ALL	12.	APU drive shaft coup torque.	bling bolts for proper						
ALL	13.	APU mounts for crac and corrosion. Remo the APU hardware, n lugs/surfaces for crac distortion or elongatio	ive APU and inspect nounts, mounting cks, corrosion, and						
ALL	14.	APU starter for crack security.	s, leakage and						
2,4	15.	APU combustor for c burned-through area							

РНА	SE NO		a Name UIPMEN	and No. NT BAY – 11		Aircraft Serial No.	Date	e
Inspect Phase Nos.	Inspection Re	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4	16. APU exhaust fairing and loose or missing							
ALL C	 APU power takeoff c continous discharge (Note: Oil stains or w acceptable.) Check f and service pro clutc 	from covered port. vetness are riction disc wear. Drain						
2,4 C	18. Change APU oil filter	Change APU oil filter.						
ALL C	19. Change APU fuel filt	er.						
ALL C	or seized bearings. F	ods. Rod ends for worn Push-pull support and security. Bellcranks , worn or seized						
ALL C	cracks, distortion, co	 Left and right engine inboard supports for cracks, distortion, corrosion, loose or missing hardware, and mounting security. 						

РНА	SEI	NO		a Name a UIPMEN	and No. NT BAY – 11		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL C	22.	Catwalk and supporti debonding, distortion hardware, and securi for peeling and deter	ty. Walk-way coating						
2,4 C	23.	Utliity accumulator tubing for leakage and security.							
ALL C	24.	Utility hydraulic manifold for leakage, corrosion, loose connections, and security. Sight gage for proper fluid level. Check manifold air inlet check valve filter for cleanliness.							
ALL C	25.	Hydraulic manifold filter indicators for popped buttons.							
ALL	26.		draulic ground service anel quick-disconnect kage.						
ALL	27.	Replace primary GSE panel fluid filter.							

РНА	PHASE NO		Area Name and No. AFT EQUIPMENT BAY – 11			Aircraft Serial No.			e
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	irks	Action Taken		Initial
ALL	28.	by activating emergency hydraulic system and operating flight controls.							
ALL	29.								

РНА	SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
	Ν	Area Name and No. /ID AND LOWER FUSELAGE – 12		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose					
		or missing hardware. Steps for damage,					
		structural integrity, and mounting security.					
ALL	2.	Access panels, fairings, and doors for					
		deformation, cracks, corrosion, loose or working rivets, and loose or missing					
		working rivets, and loose or missing hardware. Door hinges for damage, binding, and security. Latches for security and					
		proper operation. Seals and copper coat for					
		wear and deterioration. Drain holes for					
		obstructions.					
	2	Entring and motion functions outfood for					
ALL	3.	Fairing and mating fuselage surface for missing, worn, or non-adhering chafe tape.					
		missing, worn, or non-adhering chare tape.					
ALL	4.	Interior components for mounting security					
		and loose or missing hardware.					
ALL	F	- Interferente et en france de commente de la com					
ALL	5.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing					
		hardware.					

РНА	SEI	NO		a Name a DWER F	and No. TUSELAGE – 12		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL	6.	Area weapons syster hydraulic fluid leakag tubes for leakage, co Wiring harness for lo chafing, or deteriorat	e. Turret hoses and rrosion and security. ose connections,						
ALL	7.	Gun turret, gun cradle support fork, and shouldered shafts for corrosion, galling, and excessive wear.							
2,4	8.	Fuselage turret cavity for cracks, corrosion, and distortion. Inspect for bent or misaligned stringers. Gun area bulkheads for web cracks.							
ALL C	9.	cracks, corrosion, se	ght control rod and rod ends for dents, icks, corrosion, security, and evidence of erference. Rod ends for worn or seized arings.						
ALL C	10.	. Flight control bellcranks for cracks, corrosion and security. Brackets for mounting security and evidence of interference. Pivot bearings for looseness.							

РНА	SE NO	_		a Name a DWER F	and No. USELAGE – 12		Aircraft Serial No.	Dat	te
Inspect Phase Nos.	Inspec	tion Ree	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4 C	for cracks, cor	rosion,	ecoupler (ARDD) units security, evidence of ess and lost motion.						
2,4 C	security. Wirin connections, c evidence of int	LVDTS for cracks, corrosion, and mounting security. Wiring harnesses for loose connections, chafing, or deterioration and evidence of interference. Rod ends for worr and seized bearings.							
2,4 C	distortion, and	distortion, and corrosion. Check for bent or misaligned stringers.							
2,4 C	for cracks, cor								
ALL C	cracks, delami missing hardw								

РНА	SE NO		-	a Name a DWER F	and No. 'USELAGE – 12		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
2,4 C	structure	(transmissi	ast strut support on deck bottom listortion, and security.						
2,4 C	actuator deck bot	Ammo bay interior flight control servo actuator support structure (transmission deck bottom forward) for cracks, distortion, and security.							
2,4	pressure connecto	o bay interior fuel pumps and valves, sure switches, manifold, couplings, ectors and fittings for leakage, cracks, e connections and security.							
ALL	and drair corrosior coupling	n tubes for le	el lines, hoses, vent, eakage, chafing, ity. Refueling line for torque strips and						
2,4		 Ammo bay interior wiring harnesses for chafing, damaged insulation, and security. 							

РНА	SE NO		a Name a DWER F	and No. 'USELAGE – 12		Aircraft Serial No.	Dat	e
Inspect Phase Nos.	Inspection Re	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL		leakage, corrosion, mounting security, and						
2,4	for cracks, corrosion	for cracks, corrosion and loose or missing rivets. Inspect eye bolts for damage and						
2,4 C	23. Nitrogen inert compo- security, loose or mis tubes, hoses, and br damage and security obstructions.	ssing hardware. Inspect eak away values for						
ALL	24. Ammo bay panel for corrosion, and secur cracks, security, and	ity. Panel latches for						
2,4 C	25. Main landing gear cr scratches, fractures,							

РНА			a Name a OWER F	and No. USELAGE – 12	Aircraft Serial No.		Date		
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	26.	Aft stowage and avionics compartments for cleanliness, cracks, distortion, corrosion, loose or missing rivets or fastners. Seals and copper coat for deterioration and wear.							
2,4	27.	Fuselage stowage compartments for cracks, distortion, corrosion, and loose or missing rivets and hardware.							

РНА	PHASE NO PHASED MAINTENANCE CHECKLIST										
AR	MAN	Area Name and No. IENT MID AND LOWER FUSELAGE – 12		Aircraft Serial No.	Date	Total Hrs. This Area					
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial				
ALL	1.	Index drive, ejector cover, rotor, spacer, and eject guide for cracks, distortion, corrosion, nicks, gouges, and scouring, turn vertical drive shaft and check for binding.									
ALL	2.	Bolt carrier for cracks, nicks, gouges, corrosion, and scouring.									
ALL	3.	Forward track assembly for cracks, nicks, gouges, corrosion, and scouring.									
ALL	4.	Recoil adapter for leakage and physical damage. Clamp halves for corrosion, cracks, and worn clamp pin holes.									
ALL	5.	Blast suppressor for cracks, distortions and proper installation.									
ALL	6.	Barrel for cracks and distortion. Barrel support for cracks.									

РНА	SE	NO	-	a Name a	and No. WER FUSELAGE – 12		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	7.	Driver motor terminal lug for physical damage and fraying wires. Cable guard for physical damage. Wire support clamps for cracks and wear. Wire harness for chafing. Drive motor for cracks and corrosion.							
ALL	8.	Receiver assembly for cracks, distortion, burrs, nicks, gouges, and oil leakage. Turn vertical drive shaft and check for binding and corrosion.							
ALL	9.	Hydraulic drive motor tube fittings and threads for distortion, cracks, and leakage.							
ALL	10.	Hydraulic solenoid va bracket for cracks an	alve for leaks. Mounting d distortion.						
ALL	11.	Hydraulic actuator as points for cracks and							
ALL	12.	 Electrical connectors and wires for breaks, fraying and bent or broken connector pins. None allowed. 							

РНА	SEN	NO		a Name a	and No. WER FUSELAGE – 12		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	13.	broken brackets. Exc	bly for cracks, bent or essive play or binding ed drive gearbox cover.						
ALL	14.	Mounting surfaces of for cracks and distort							
ALL	15.	 Azimuth resolver assembly adapter and spur gear for cracks and distortions. 							
ALL	16.	Elevation resolver as cracks and distortion							
ALL	17.	 Adapter mounting holes in trunnion shaft for cracks and stripped threads. 							
ALL	18.	Train rate sensor morand distortions.	unting point for cracks						
ALL	19.	 Turret wiring and connections for breaks, fraying, and broken or bent pins. 							

PHA	SEI	NO		a Name a	and No. WER FUSELAGE – 12		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL	20.	Turret mounting area corrosion.	for cracks and						
ALL	21.	Support cradle for be	ends and distortions.						
ALL	22.	Trunnion bearing for	damage.						
ALL	23.	. Support fork for damage, cracks, deep scratches in either leg and corrosion, damaged trunnion and threaded inserts.							
ALL	24.	Gun turret gun cradle support fork shouldered shafts for corrosion, galling and excessive wear.							
ALL	25.	Fork mounting area on azimuth housing and cradle for cracks, distortion, and stripped threads.							
ALL	26.	Stow spring assembl cracks, distortion, an	y mounting points for d corrosion.						

РНА	SEN	NO		a Name a AND LO'	and No. WER FUSELAGE – 12		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL	27.	Ammo storage maga damage, and corrosid	zine mounts for cracks, on.						
ALL	28.	Magazine for loose o fastners, screws, cra							
ALL	29.	Accelerator/merger assembly for cracks, loose or missing hardware, damage and corrosion.							
ALL	30.	. Carrier drive assembly left and right flex chutes for cracks, damage and corrosion.							
ALL	31.	Carrier drive assemb hoses for damage an							
ALL	32.	Carrier drive assemb corrosion and bent or	ly connector for cracks, r broken pins.						
ALL	33.	Carrier drive servo manifold for cracks and damage.							

РНА	SEI	NO		a Name a	and No. WER FUSELAGE – 12		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL	34.	attach bracket, brack	et assembly, upper and for cracks, distortions,						
ALL	35.	Ammo conveyor syst corrosion and proper							
ALL	36.	5. Remove conveyor assembly from aircraft. Check all feeders for cracks, bends, and elongated connecting holes. None allowed. Use ammunition tool kit.							
ALL	37.	 Area weapon system for evidence of hydraulic fuild leakage. Turret hoses and tubes for leakage, corrosion, and security. Wiring harnesses for loose connections, chafing and deterioration. 							
ALL	38.	Aft avionics equipment for mounting security, external damage, loose connectors. Wiring harnesses and clamps for chafing and deterioration.							

РНА	SE	NO P	HASED	MAINTENANCE CHECKLIST			
		Area Name and No. TAILBOOM – 13		Aircraft Serial No.	Date	Total Hrs. This Are	a
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	A	ction Taken	Initial
ALL	1.	Exterior skin for holes, dents, cracks, corrosion, loose or working rivets, and loose or missing hardware. Steps for damage, structural integrity, and mounting security.					
ALL	2.	Access panels and fairings for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Fairing, hinges and latches for damage, binding, security, and proper operation. Seals and copper coat for wear or deterioration.					
ALL	3.	Interior components for mounting security and loose or missing hardware.					
ALL	4.	Tailboom deck structure for cracks, distortion, corrosion, loose or working rivets, and loose or missing hardware.					

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

РНА	SEI	NO		Area Name and No. TAILBOOM – 13			Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4 C	5.	Tail rotor drive shaft and couplings for cracks, dents, distortion, corrosion, and evidence of interference.							
2,4 C	6.	Tail rotor drive shaft and coupling bolts for proper torque to 125 in. lbs.							
2,4 C	7.	Drive shaft dampers for proper friction adjustment.							
ALL C	8.	Tail rotor flight contro cracks, corrosion, se interference. Check r seized bearings.	curity, and evidence of						
ALL C	9.	Tail rotor flight contro corrosion, security, a interference. Pivot be							

РНА	SE NO.		Area Name and No. TAILBOOM – 13			Aircraft Serial No.			e
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	rks	Action Taken		Initial
2,4	line	 Tailboom interior hydraulic components and lines for leakage, dents, corrosion, chafing and security. 							
2,4		. Tailboom interior wiring harnesses for chafing, deterioration, and security.							
2,4		lboom armor chan se or missing harc	nels for cracks and dware.						
2,4 C	she	Tailboom splice (FS 436.5/476.6) for sheared or working rivets, corrosion, and cracked or deformed skin.							

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

PHA	PHASE NO PHASED MAINTENANCE CHECKLIST										
E	MPE	Area Name and No. ENNAGE, TAIL ROTOR, AND TLG – 14	Aircraft Serial No.		Date	Total Hrs. This Area	-				
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial				
ALL	1.	Exterior skin for holes, dents, cracks, corrosion, loose or working rivets, and loose or missing hardware. Steps for damage, structural integrity, and mounting security.									
ALL	2.	Access panels and fairings for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Lanyards for security and condition. Check for worn or deteriorated seals and copper coat.									
ALL	3.	Structural flanges and fairings on vertical stabilizer for worn, non-adhering, or missing chafe tape.									
ALL	4.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.									

PHA	SE	NO	-	a Name a AIL ROT	and No. FOR, AND TLG – 14		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4 C	5.	Vertical stabilizer mount fittings for cracks and distortion. Mounting bolts and barrel nuts for damage, security, and proper installation.							
2,4 C	6.	Stabilizer navigation and formation lights for corrosion, loose or missing hardware, and security. Lenses for cracks, security, and discoloration. Wiring harnesses for loose connections and chafing. (Remove R578).							
2,4	7.	antenna, and FM-AM whip antenna for damage, mounting security, and condition of wiring. (Remove T545). Stabilizer spar box for cracks, corrosion, distortion, and loose or working rivets. (Access L530, L550).							
2,4 C	8.								
2,4 C	9.								

РНА	SE I	NO		a Name a AIL ROT	and No. FOR, AND TLG – 14		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
2,4 C	10.	Stabilizer trailing edg and corrosion.	Stabilizer trailing edge fairing for damage and corrosion.						
ALL C	11.		Stabilizer structure and skin for cracks, dents, distortion, and loose or missing rivets and hardware.						
ALL C	12.	Stabilator tip fairings for cracks and loose or missing screws.							
ALL C	13.	and corrosion. Wiring for chafing, deterioration, and connection security. Stabilator and aircraft fittings for cracks, loose or missing hardware, and worn or seized bearings.							
ALL C	14.								

РНА	SEI	NO		a Name a AIL RO	and No. FOR, AND TLG – 14		Aircraft Serial No.	Date	e
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL C	15.	Stabilator pivot bolts, pivot bearings, and pivot bosses for cracks, distortion, corrosion, and cleanliness. Pivot bearings for looseness. (Remove stabilator).							
ALL C	16.	Intermediate gearbox housing, input and output retainers for cracks, distortion, and security. Check for grease leakage and evidence of overheating (discoloration). Gearbox mount fittings for cracks and distortion.							
2,4 C	17.	 proper torque. Intermediate gearbox cooling fins, shroud, and defuser for cracks and distortion. Impeller for cracks and loose or missing hardware. 							
ALL C	18.								
2,4	19.								

РНА	SE I	NO		a Name AIL RO	and No. TOR, AND TLG – 14		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4	20.	 Intermediate gearbox wiring harness for loose connections, chafing, or deterioration. 							
2,4 C	21.	 Tail rotor drive shaft and couplings for cracks, dents, distortion, and corrosion. 							
2,4 C	22.	. Tail rotor drive shaft and coupling bolts, verify torque.							
ALL C	23.	 seized bearings, cracked or broken fittings, cracked strut, and loose or missing hardware. 4. Tail rotor gearbox housing, and input and output retainers for cracks, distortion, security. Check for grease leakage and evidence of overheating (discoloration). 							
ALL C	24.								
ALL C	25.								

РНА	SEI	NO	-	a Name a AIL ROT	and No. ΓOR, AND TLG – 14		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	nrks	Action Taken		Initial
ALL C	26.	Tail rotor gearbox mounting studs, verify torque.							
2,4	27.	Tail rotor gearbox thermistors for insulation damage, grease leakage, and security. Wire harness splices for security, loose connections, chafing and deterioration.							
ALL C	28.	Tail rotor flight control rods for dents, cracks. corrosion, security, and evidence of interference. Check rod ends for worn or seized bearings.							
ALL C	29.	Tail rotor pitch change links for dents, cracks, corrosion, security, evidence of interference, worn or seized bearings, and tolerance wear limits.							
2,4 C	30.	Tail rotor flight control bellcranks for cracks, corrosion, security, and evidence of interference. Pivot bearings for looseness.							

РНА	SEI	NO		a Name and AIL ROTO	d No. DR, AND TLG – 14		Aircraft Serial No.	Date	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4 C	31.	 leakage, cracks, and corrosion. Rod end and base clevis for bearing damage and attachment security. Rod end for looseness. Rubber boot for cuts, tears, and deterioration. Inspect servo cylinder control linkage hardware for damage and security. Tail rotor swashplates for cracks and corrosion. Looseness between stationary and rotating swashplates. Check -901 swashplate. Tail rotor head for cracks, distortion, corrosion, and security. Attaching studs for looseness. Check free play measurement on -9/-13 tail rotor swashplates. 							
2,4 C	32.								
ALL C	33.								
ALL C	34.								

PHA	SEI	NO	-	a Name a AIL ROT	and No. ΓOR, AND TLG – 14		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
2,4	35.	 studs from the tail rotor gearbox output shaft for wear and three nuts found on those studs for torque. Torque to 60 foot-pounds. Tail rotor root blade bolt heads and nuts for cracks and looseness. (PLI Method) Tail rotor blade spars and root finger doublers for delamination. Tail rotor blade leading edge for cracks, dents, distortion, and erosion. Leading edge tips for loose or missing hardware. 							
ALL C	36.								
ALL C	37.								
ALL C	38.								
2,4 C	39.								

РНА	SEI	NO		a Name a AIL RO	and No. ΓOR, AND TLG – 14		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
2,4 C	40.	Tail landing gear arms for cracks, distortion, and security. Pivot pin, bolts, and bushings for looseness.							
ALL C	41.	. Tail landing gear wheel lock actuator for leakage, cracks, and security. Check locking handle and linkage for free movement, full travel, and lateral looseness.							
2,4 C	42.	Tail landing gear wiring harness for loose connections, chafing, or deterioration.							
2,4 C	43.	Repack Tail Landing	Gear wheel bearings.						
ALL C	44.	 Tail landing gear shock strut for leakage, cracks, distortion, and corrosion. Rod ends for worn or seized bearings. 							
2,4 C	45.	Tail boom end frame (F.S. 547.15) for cracks, corrosion, worn or seized bearings and bushings, loose or working rivets.							

РНА	PHASE NO PHASED MAINTENANCE CHECKLIST										
POST	Area Name CONDITIONS AND P	and No. OWER ON CHECKS – 15		Aircraft Serial No.	Date	Total Hrs. This Area					
Inspect Phase Nos.	Inspectio	n Requirements	Status	Faults and/or Remarks		Action Taken	Initial				
ALL	1. Install mast mou	nted assembly. (if required).									
ALL	2. Install main rotor	blades.									
ALL	3. Install conveyor	assembly.									
ALL	4. Install gun turret	assembly.									
ALL	5. Install ammo ma	gazine.									
ALL	6. Lubricate ammu	nition handling subsystem.									
ALL	7. Adjust ammunitio	on handling subsystem.									
ALL	8. Repanel aircraft										
ALL	9. Install store jettis	on cartridges.									

РНА	SEN	NO		a Name a AND PC	and No. WER ON CHECKS – 15		Aircraft Serial No.	Date	9
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Remark	S	Action Taken		Initial
ALL	10.	Start APU.							
с									
ALL	11.	Perform ARDD 250 h	nour inspection.						
с									
ALL C	12.	 Flight controls actuated through full stop-to-stop ranges. Main and tail rotor pitch change functions for full travel, control response, binding, looseness and noisy operation. 							
2,4 C	13.	Perform battery char	ger operational check.						
ALL C	14.	Operate fuel boost pu light.	ump for fuel pressure						
ALL C	15.	15. Perform fuel system leak-check.							
ALL C	16.	 Stabilator actuated through full travel range for smooth operation with no lost motion or binding. 							

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

РНА	SEI	NO		a Name a AND PC	and No. DWER ON CHECKS – 15		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	irks	Action Taken		Initial
ALL C	17.	Forward and aft avionics bay cooling fans for smooth operation.							
ALL C	18.	 Perform windshield anti-ice system operational check. 							
ALL C	19.	 Operate rotor brake for brake, lock, and off operational modes. 							
ALL C	20.	Perform pitot heat ch	eck.						
ALL C	21.	Perform tail rotor and alarm 250 hour inspe	d intermediate gearbox action						
ALL C	22.	2. <u>WARNING:</u> DO NOT FIRE LASER. PERFORM TADS/PNVS OPERATIONAL CHECK FOR SMOOTH COMPONENT FUNCTIONS.							
ALL C	23.	Start both engines.							

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspection.

РНА	SEI	NO	-	a Name a AND PC	and No. DWER ON CHECKS – 15		Aircraft Serial No.	Date	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Remar	ks	Action Taken		Initial
ALL C	24.	 Perform HIT checks on both engines. 							
ALL C	25.	 Perform FD/LS check and MOC for main transmisson and nose gearbox transducers, temperature probes, and oil pressure transducers. 							
ALL C	26.	. Operate engine anti-ice system. Verify rise in engine TGT.							
ALL C	27.	Check ECS for adeq maximum heat and c and CPG stations.	uate air flow and old outputs in both pilot						
ALL C	28.	Perform engine chop operational check. TM 1-1520-251-MTF.							
ALL C	29.	29. Ensure that all entries on forms, records, and work sheets have been completed or updated and new forms initiated as required, and/or have been carried forward on a DA Form 2408-13 or DA Form 2408-14 in accorance with DA Pam 738-751.							

РНА			a Name a AND PO	and No. WER ON CHECKS – 15		Aircraft Serial No.	Dat	e	
Inspect Phase Nos.		Inspection Requirements			Faults and/or Rema	rks	Action Taken		Initial
ALL C		D. Perform post-inspection maintenance operational checks (MOC). After the completion of any required corrective actions to any of the components of a functional system of the aircraft. MOCs shall be performed on the system to determine the effectiveness of the maintenance actions performed and to verify the proper operation of the system. These MOC's shall be performed in accordance with the IETM. Copies of supplemental sheets (DA Form 4676-R) may be used to record and sign off the maintenance operational checks performed.							
ALL C		 Perform post-inspection maintenance operational checks (MOC), as required. Perform MOC's on aircraft systems or components that have been disturbed during the inspection. MOC's shall be performed in accordance with the IETM. 							
ALL C	32.	2. Perform 10 hour/14 day inspection.							

РНА	PHASE NO POST			a Name a AND PC	and No. WER ON CHECKS – 15		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Remar	ks	Action Taken		Initial
ALL C	33.	 Perform post-inspection MTF in accordance with TM 1-1520-251-MTF and TM 1-1520-328-23. 							
ALL C	34.	 Final records check for completion of AH-64D helicopter phased maintenance inspection. 							

PHA	PHASE NO PHASED MAINTENANCE CHECKLIST										
	FIN	Area Name and No. NAL INSPECTION REQUIREMENTS		Aircraft Serial No.	Date	Total Hrs. This Area					
Inspect Phase Nos.	t Inspection Requirements			Faults and/or Remarks		Action Taken	Initial				
ALL C	1.	Ensure that all entries on forms, records, and work sheets have been completed or updated and new forms initiated as required, and/or have been carried forward on DA Form 2408-13 or DA Form 2408-14 in accordance with DA PAM 738-751.									
ALL C	2.	Perform Post-Inspection Maintenance Operational Checks (MOC), as required, in accordance with requirements of TM 1-1500-328-23.									
ALL C	3.	Perform 10 Hour/14 Day Inspection in accordance with IETM.									
ALL C	4.	Perform Post-Inspection MTF in accordance with TM1-1520-251-MTF and TM 1-1500-328-23.									

By Order of the Secretary of the Army:

OFFICIAL:

Joel B. Hub

JOEL B. HUDSON Administrative Assistant to the Secretary of the Army 0210101

ERIC K. SHINSEKI General, United States Army Chief of Staff

DISTRIBUTION:

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THESE ARE THE INSTRUCTIONS FOR SENDING AN ELECTRONIC 2028

The following format must be used if submitting an electronic 2028. The subject line must be exactly the same and all fields must be included; however only the following fields are mandatory: 1, 3, 4, 5, 6, 7, 8, 9, 10, 13, 15, 16, 17, and 27.

- From: "Whomever" <whomever@avma27.army.mil>
- To: 2028@redstone.army.mil
- Subject: DA Form 2028
 - 1. From: Joe Smith
 - 2. Unit: Home
 - 3. Address: 4300 Park
 - 4. City: Hometown
 - 5. **St:** MO
 - 6. **Zip:** 77777
 - 7. Date Sent: 19-Oct-93
 - 8. **Pub no:** 55-2840-229-23
 - 9. Pub Title: TM
- 10. **Publication Date:** 04-Jul-85
- 11. Change Number: 7
- 12. Submitter Rank: MSG
- 13. Submitter Fname: Joe
- 14. Submitter Mname: T
- 15. **Submitter Lname:** Smith
- 16. **Submitter Phone:** (123) 123-1234
- 17. **Problem:** 1
- 18. Page: 2
- 19. Paragraph: 3
- 20. Line: 4
- 21. NSN: 5
- 22. Reference: 6
- 23. Figure: 7
- 24. Table: 8
- 25. Item: 9
- 26. Total: 123
- 27. **Text:**

This is the text for the problem below line 27.

			THEN. DOPE A FORM.	METHING	FROM: (P SSG E. Tro	rint your unit's complete address) John Doe oop 5th Cav. 1st Training Building Knox, Kentucky 12345-6789
			IN THE	OLD IT AND DROP IT MAIL.		21 June 1994
PUBLICATI	ON NUMBER	520-251	1-PM			PUBLICATION TITLE Phased Maintenance Inspection Checklist for AH-64D Helicopter
PAGE NO. 2-4	PIN–POIN PARA- GRAPH	T WHERE IT IS FIGURE NO.	TABLE NO.	IN THIS SPACE TELL WHAT IS AND WHAT SHOULD BE DONE	about it: ut insp	ection areas 2 and 4. ALL is column.
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SP4		or title, and Doe, Auto	ovon 5			PSIF YOUR OUTFIT WANTS TO KNOW ABOUT RECOMMENDATIONS MAKE A CARBON COPY OF THIS AND GIVE IT TO YOUR HEADQUARTERS

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	PUBLICAT						DATE SENT	PUBLICATIO	DN TITLE	
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			OR TITLE, AND		NUMBER	PREVIOUS ARE OB:		F	PS IF YOUR OUTFIT WA RECOMMENDATIONS MAK HIS AND GIVE IT TO YOU	



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DEPARTMENT OF THE ARMY

OFFICIAL BUSINESS PENALTY FOR PRIVATE USE, \$300 POSTAGE AND FEES PAID FOR DEPARTMENT OF THE ARMY DOD 314



TEAR ALONG PERFORATED LINE

Commander U.S. Army Aviation And Missile Command ATTN: AMSAM–MMC–MA–NP Redstone Arsenal, AL 35898–5230

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Commander U.S. Army Aviation And Missile Command ATTN: AMSAM–MMC–MA–NP Redstone Arsenal, AL 35898–5230

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EXACT AGE NO.	PARA- GRAPH	T WHERE IT IS FIGURE NO.	TABLE NO.		CE TELL WHAT IS SHOULD BE DONE		
	ME, GRADE	OR TITLE, AND	TELEPHONE	NUMBER		SIGN HERE	



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The Metric System and Equivalents

Linear Measure

- 1 centimeter = 10 millimeters = .39 inch
- 1 decimeter = 10 centimeters = 3.94 inches
- 1 meter = 10 decimeters = 39.37 inches
- 1 dekameter = 10 meters = 32.8 feet
- 1 hectometer = 10 dekameters = 328.08 feet
- 1 kilometer = 10 hectometers = 3,280.8 feet

Weights

- 1 centigram = 10 milligrams = .15 grain
- 1 decigram = 10 centigrams = 1.54 grains
- 1 gram = 10 decigrams = .035 ounce
- 1 dekagram = 10 grams = .35 ounce
- 1 hectogram = 10 dekagrams = 3.52 ounces
- 1 kilogram = 10 hectograms = 2.2 pounds
- 1 quintal = 100 kilograms = 220.46 pounds 1 metric ton = 10 quintals = 1.1 short tons

Liquid Measure

- 1 centiliter = 10 milliliters = .34 fl. ounces
- 1 deciliter = 10 centiliters = 3.38 fl. ounces
- 1 liter = 10 deciliters = 33.81 fl. ounces
- 1 dekaliter = 10 liters = 2.64 gallons
- 1 hectoliter = 10 dekaliters = 26.42 gallons
- 1 kiloliter = 10 hectoliters = 264.18 gallons

Square Measure

- 1 sq. centimeter = 100 sq. millimeters = .155 sq. inch
- 1 sq. decimeter = 100 sq. centimeters = 15.5 sq. inches
- 1 sq. meter (centare) = 100 sq. decimeters = 10.76 sq. feet
- 1 sq. dekameter (are) = 100 sq. meters = 1,076.4 sq. feet
- 1 sq. hectometer (hectare) = 100 sq. dekameters = 2.47 acres
- 1 sq. kilometer = 100 sq. hectometers = .386 sq. mile

Cubic Measure

- 1 cu. centimeter = 1000 cu. millimeters = .06 cu. inch
- 1 cu. decimeter = 1000 cu. centimeters = 61.02 cu. inches
- 1 cu. meter = 1000 cu. decimeters = 35.31 cu. feet

Approximate Conversion Factors

To change	То	Multiply by	To change	То	Multiply by
inches	centimeters	2.540	ounce-inches	newton-meters	.007062
feet	meters	.305	centimeters	inches	.394
yards	meters	.914	meters	feet	3.280
miles	kilometers	1.609	meters	yards	1.094
square inches	square centimeters	6.452	kilometers	miles	.621
square feet	square meters	.093	square centimeters	square inches	.155
square yards	square meters	.836	square meters	square feet	10.764
square miles	square kilometers	2.590	square meters	square yards	1.196
acres	square hectometers	.405	square kilometers	square miles	.386
cubic feet	cubic meters	.028	square hectometers	acres	2.471
cubic yards	cubic meters	.765	cubic meters	cubic feet	35.315
fluid ounces	milliliters	29.573	cubic meters	cubic yards	1.308
pints	liters	.473	milliliters	fluid ounces	.034
quarts	liters	.946	liters	pints	2.113
gallons	liters	3.785	liters	quarts	1.057
ounces	grams	28.349	liters	gallons	.264
pounds	kilograms	.454	grams	ounces	.035
short tons	metric tons	.907	kilograms	pounds	2.205
pound-feet	newton-meters	1.356	metric tons	short tons	1.102
pound-inches	newton-meters	.11296			

Temperature (Exact)

$^{\circ}$ F	Fahrenheit	5/9 (after	Celsius
	temperature	subtracting 32)	temperatu

 $^{\circ}$ C

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PIN: 073278-000